

# **L9: Lagrangian Dynamics**

**Hao Su**

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*The flow and some contents are based on ECE5463 taught at Ohio State University by Prof. Wei Zhang*

# Agenda

- Lagrangian Method
- Example: Inverted Pendulum
- Example: Cart Pole
- Example: Single-Object Dynamics
- Example: Robot Arm

click to jump to the section.

# Dynamics Example: Grasp

- Consider the right grasp problem

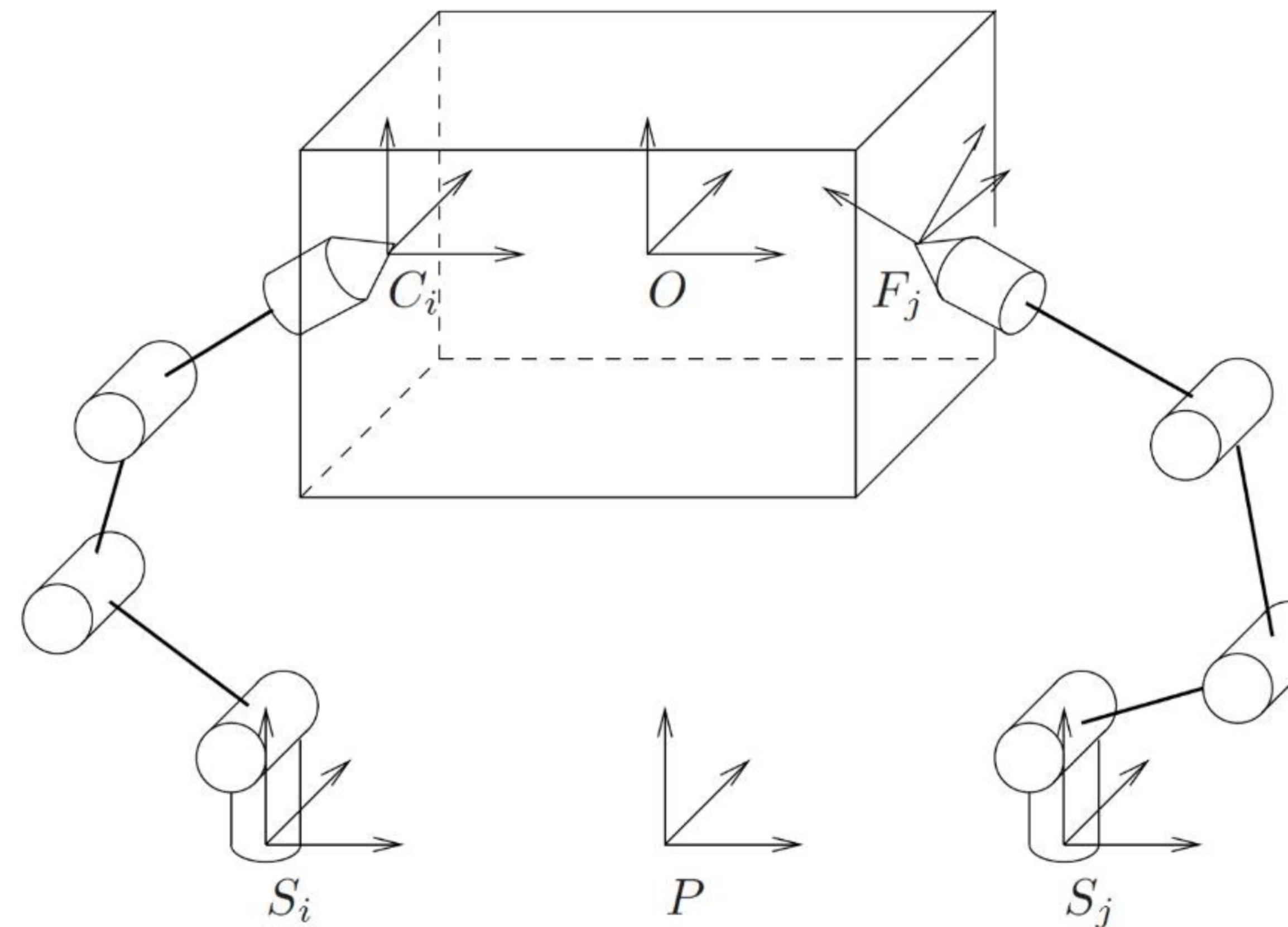


Figure 5.14: Grasp coordinate frames.

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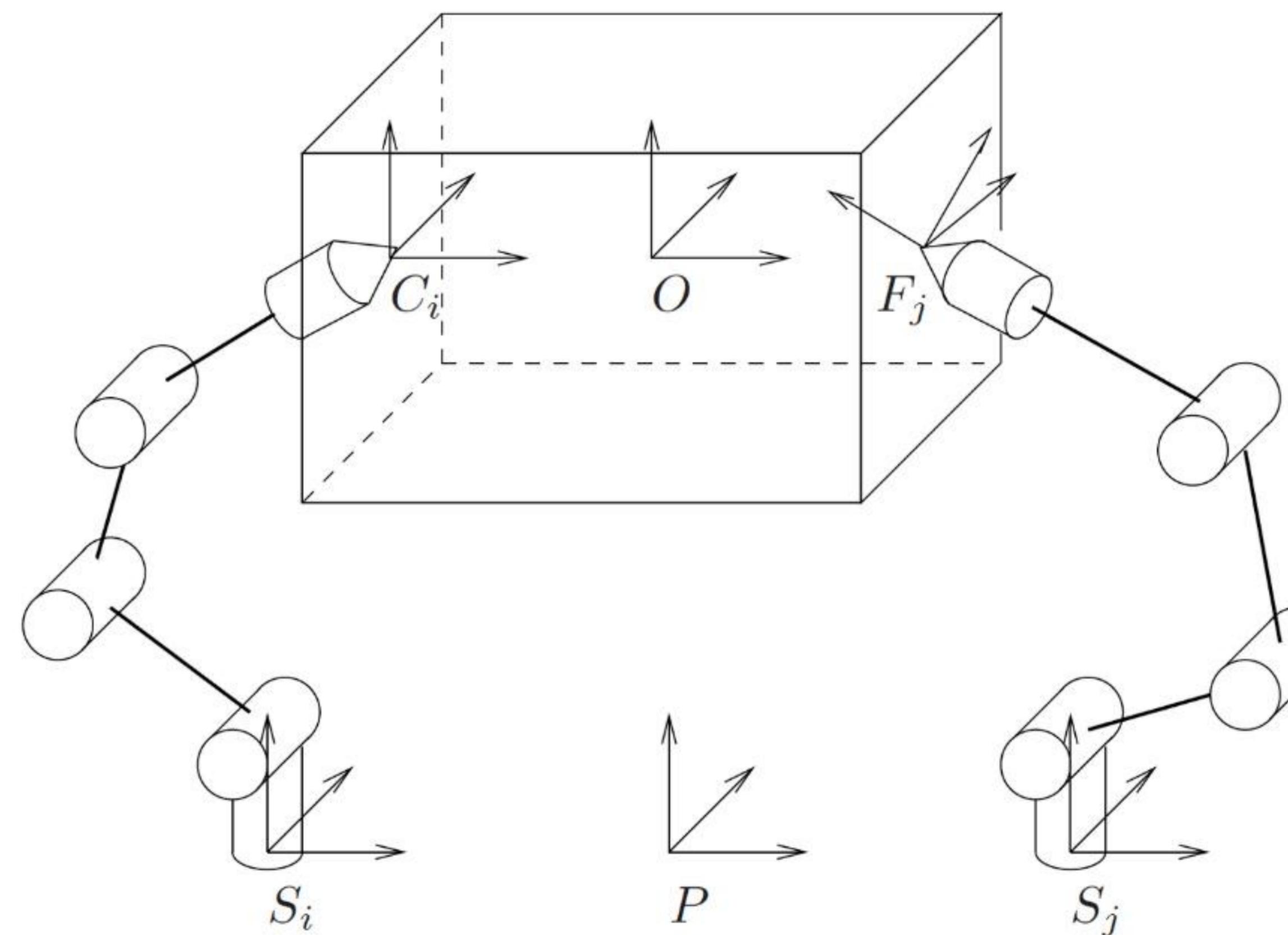


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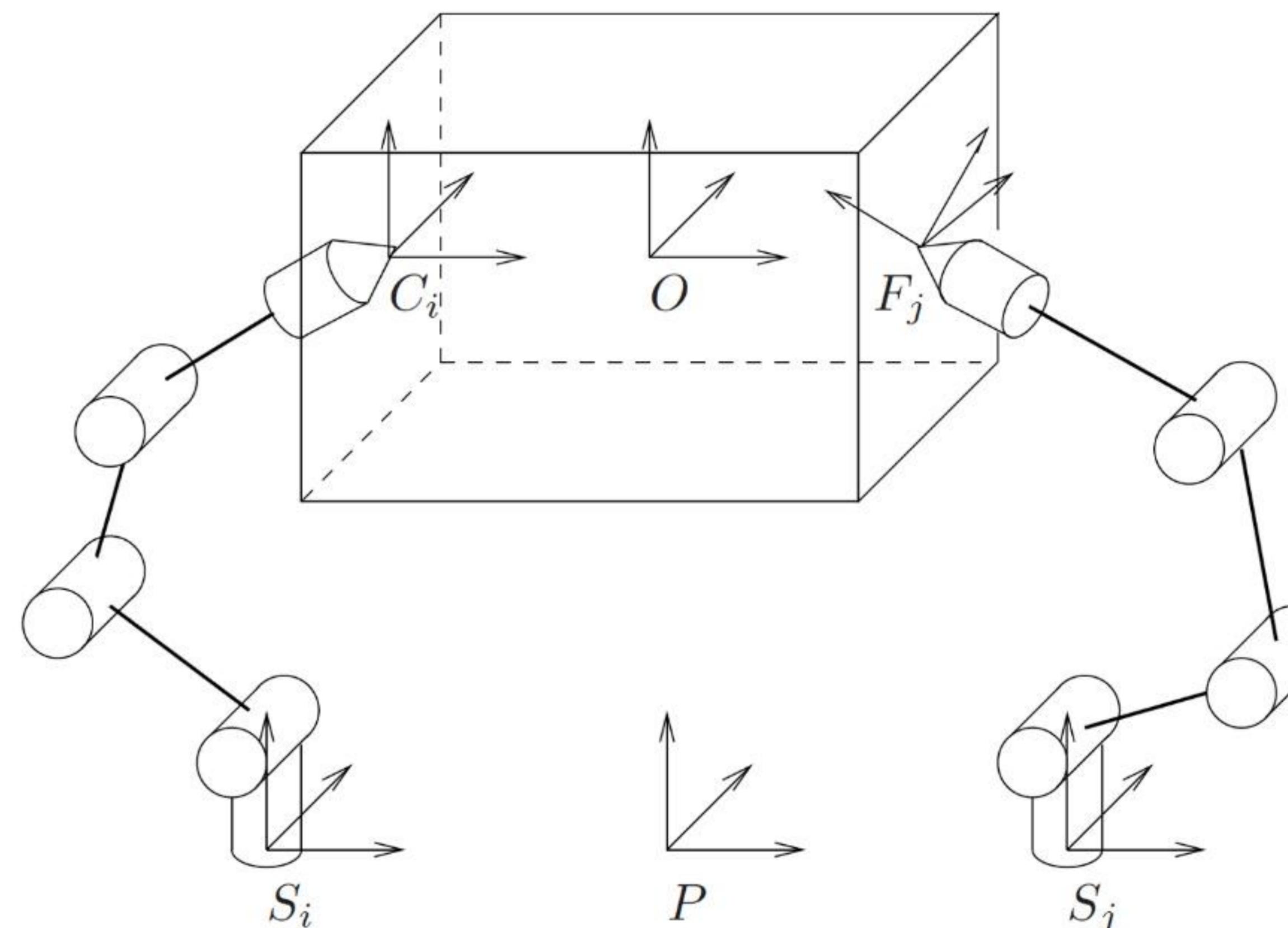


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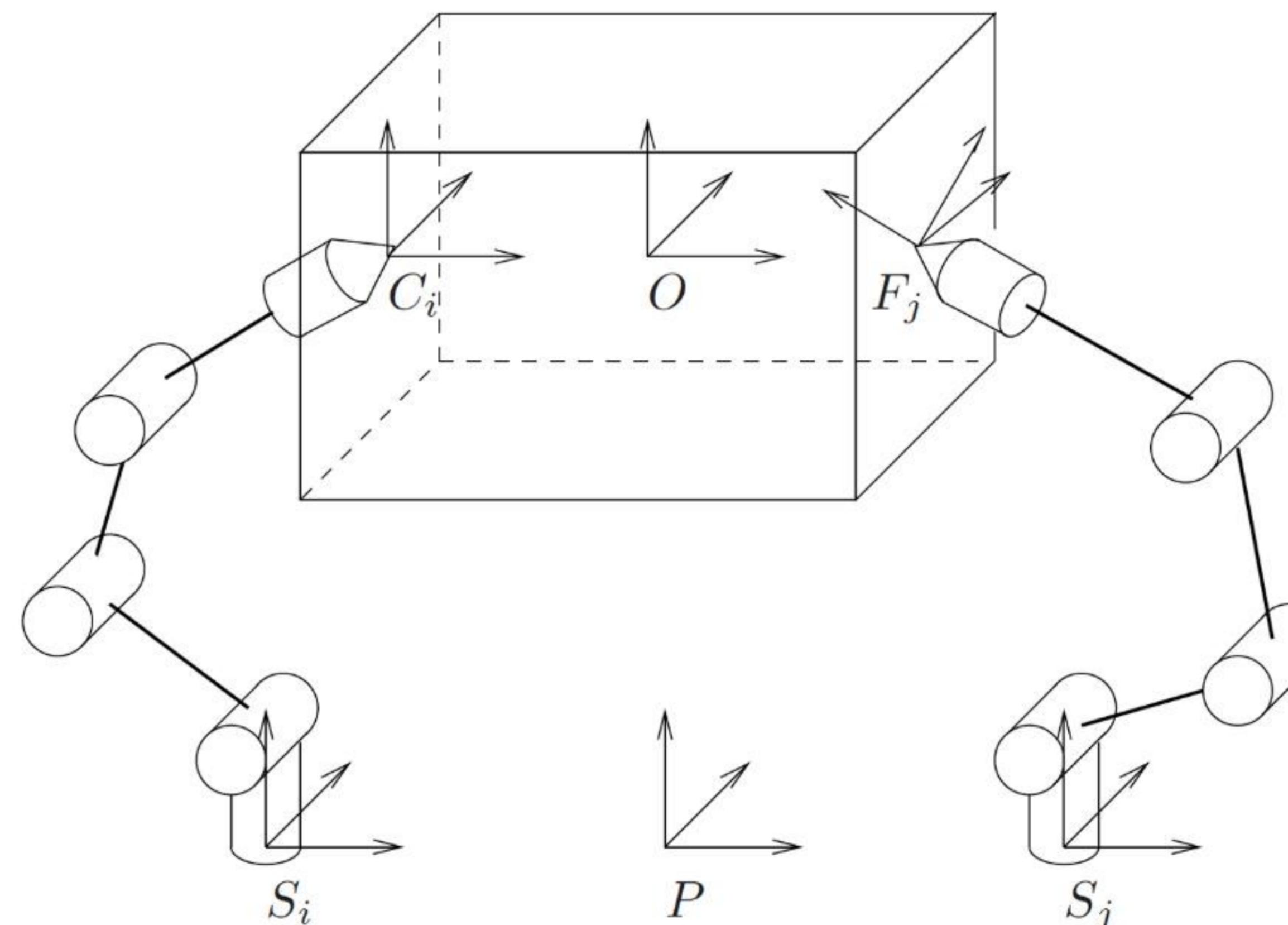


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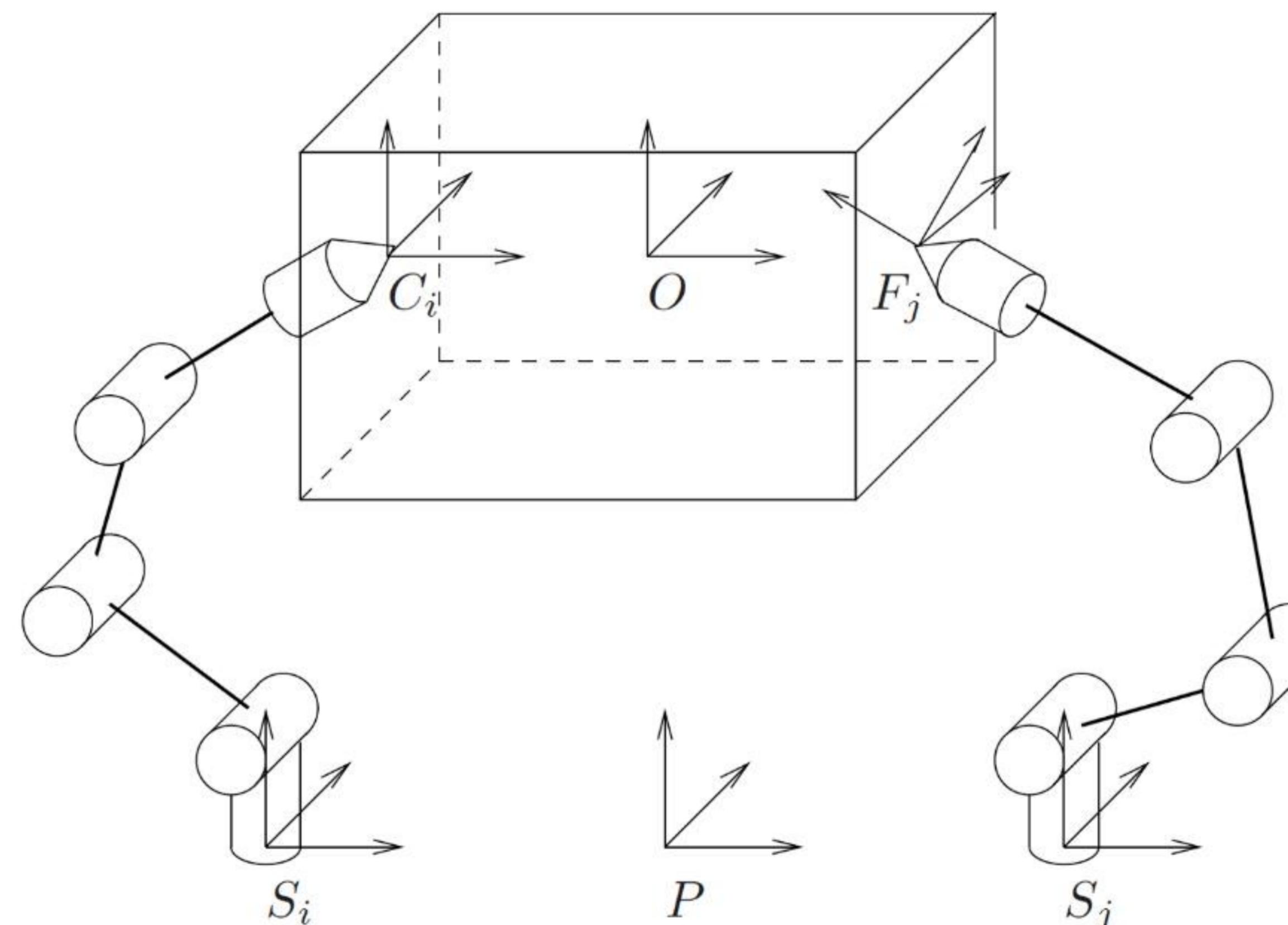


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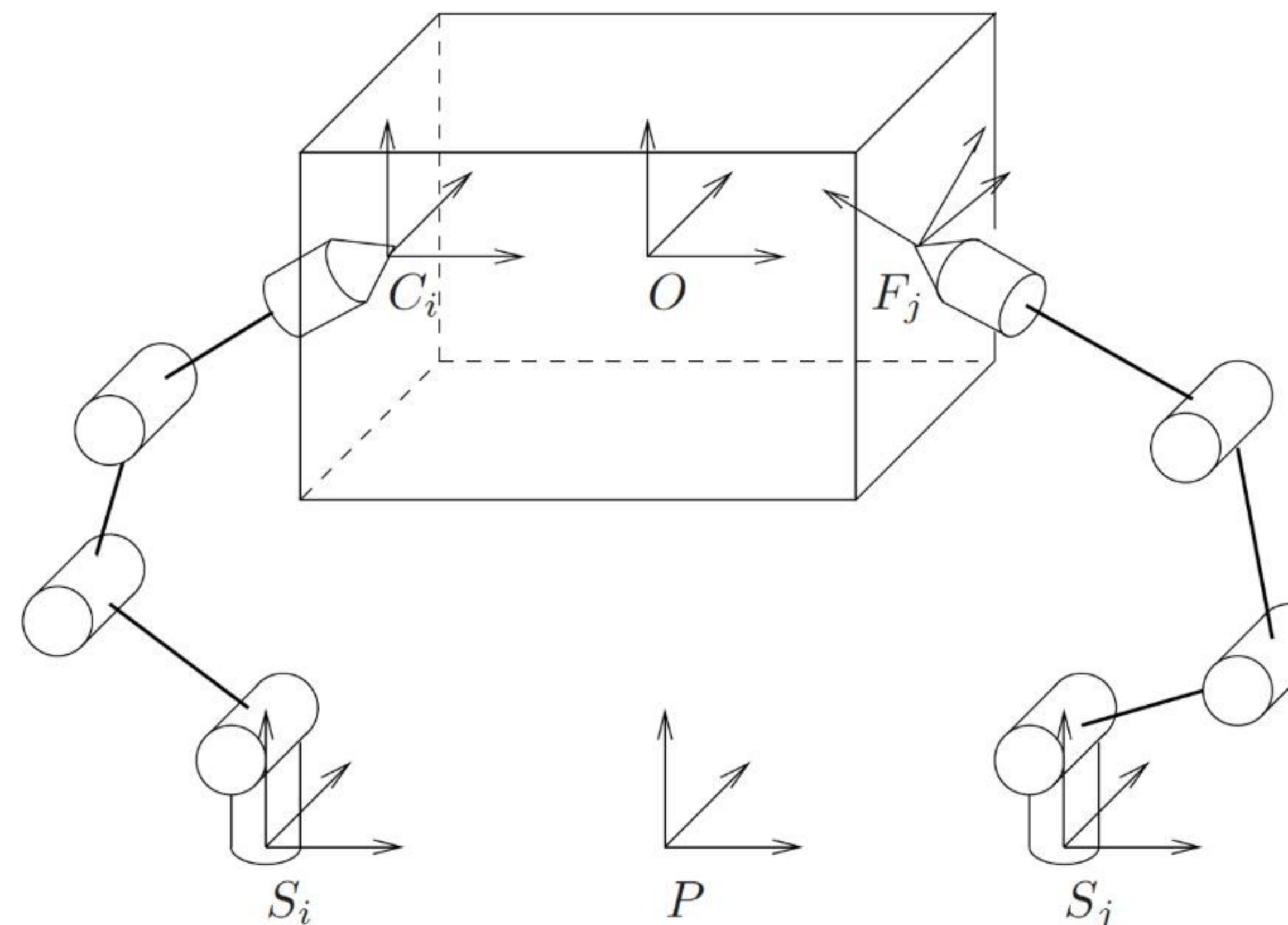


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**Q1: How to compute force at the tips from the torques at joints?**

**Q2: To keep the box static, what is the balance condition?**

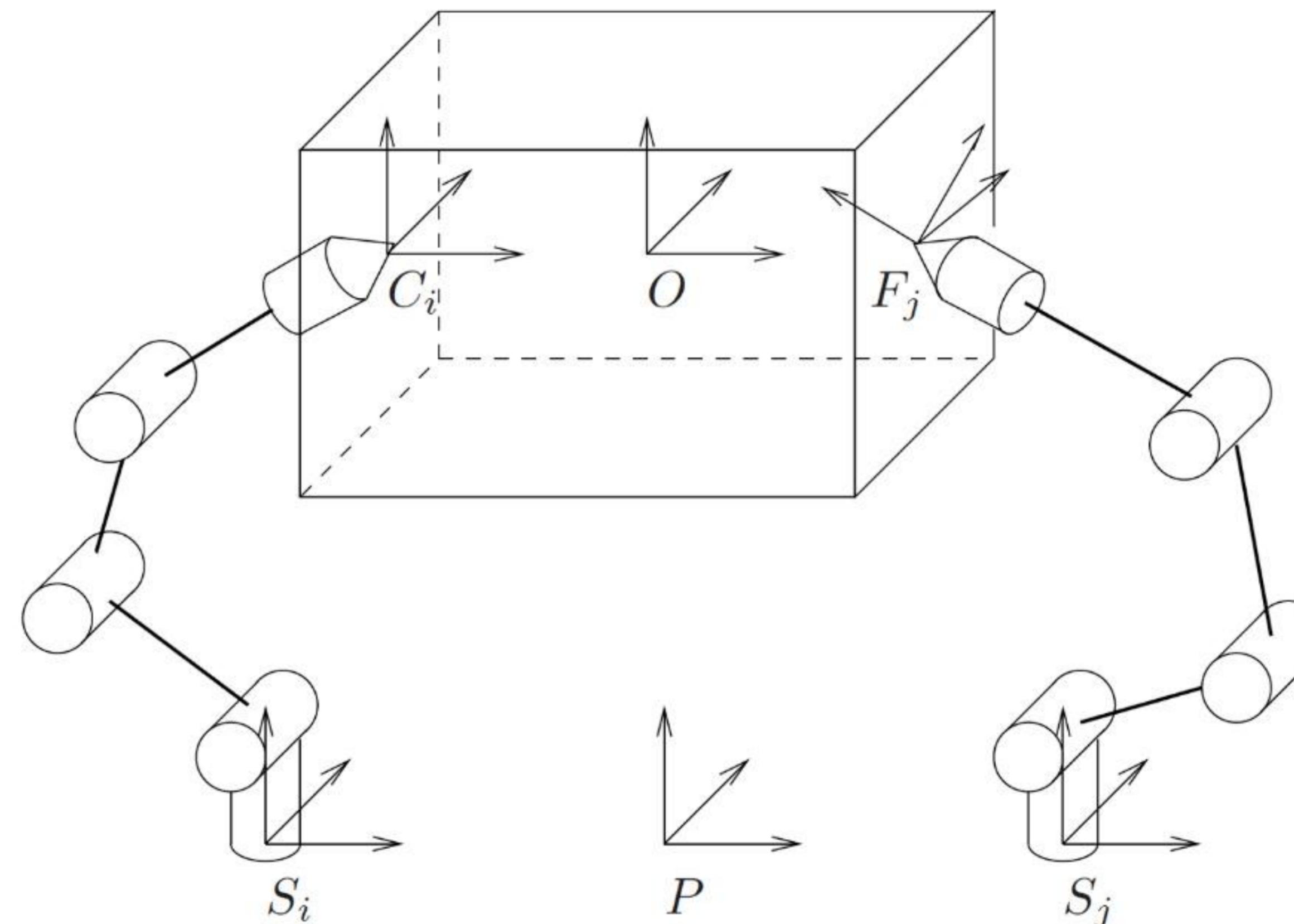


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# Dynamics Example: Grasp

- Parameterization
  - $\theta \in \mathbb{R}^n$ : vector of joint variables
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- Parameterization
  - $\theta \in \mathbb{R}^n$ : vector of joint variables
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- Task
  - **Forward dynamics:** Determine acceleration  $\ddot{\theta}$  given the state  $(\theta, \dot{\theta})$  and the joint forces/torques

$$\ddot{\theta} = \text{FD}(\tau; \theta, \dot{\theta})$$

- **Inverse dynamics:** Finding torques/forces given state  $\theta, \dot{\theta}$  and desired acceleration  $\ddot{\theta}$

$$\tau = \text{ID}(\ddot{\theta}; \theta, \dot{\theta})$$

# Lagrangian vs. Newton-Euler Methods

- There are typically two ways to derive the equation of motion for an open-chain robot: Lagrangian method and Newton-Euler method

## Lagrangian Formulation

- Energy-based method
- Often used for study of dynamic properties and analysis of control methods

## Newton-Euler Formulation

- Balance of forces/torques
- Often used for numerical solution of forward/inverse dynamics

# Lagrangian Method

# Generalized Coordinates and Forces

- Consider  $k$  particles. Let  $\mathbf{f}_i$  be the force acting on the  $i$ th particle,  $\mathbf{p}_i$  be its position.

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- Now consider the case in which some particles are rigidly connected, imposing constraints on their positions

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- $k$  particles in  $\mathbb{R}^3$  under  $n_c$  constraints  $\Rightarrow 3k - n_c$  degree of freedom
- We introduce  $n := 3k - n_c$  independent variables  $q_i$ 's, called the **generalized coordinates**

$$\begin{cases} \alpha_j(p_1, \dots, p_k) = 0 \\ j = 1, \dots, n_c \end{cases} \quad \Leftrightarrow \quad \begin{cases} p_i = \gamma_i(q_1, \dots, q_n) \\ i = 1, \dots, k \end{cases}$$

# Generalized Coordinates and Forces

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# Generalized Coordinates and Forces

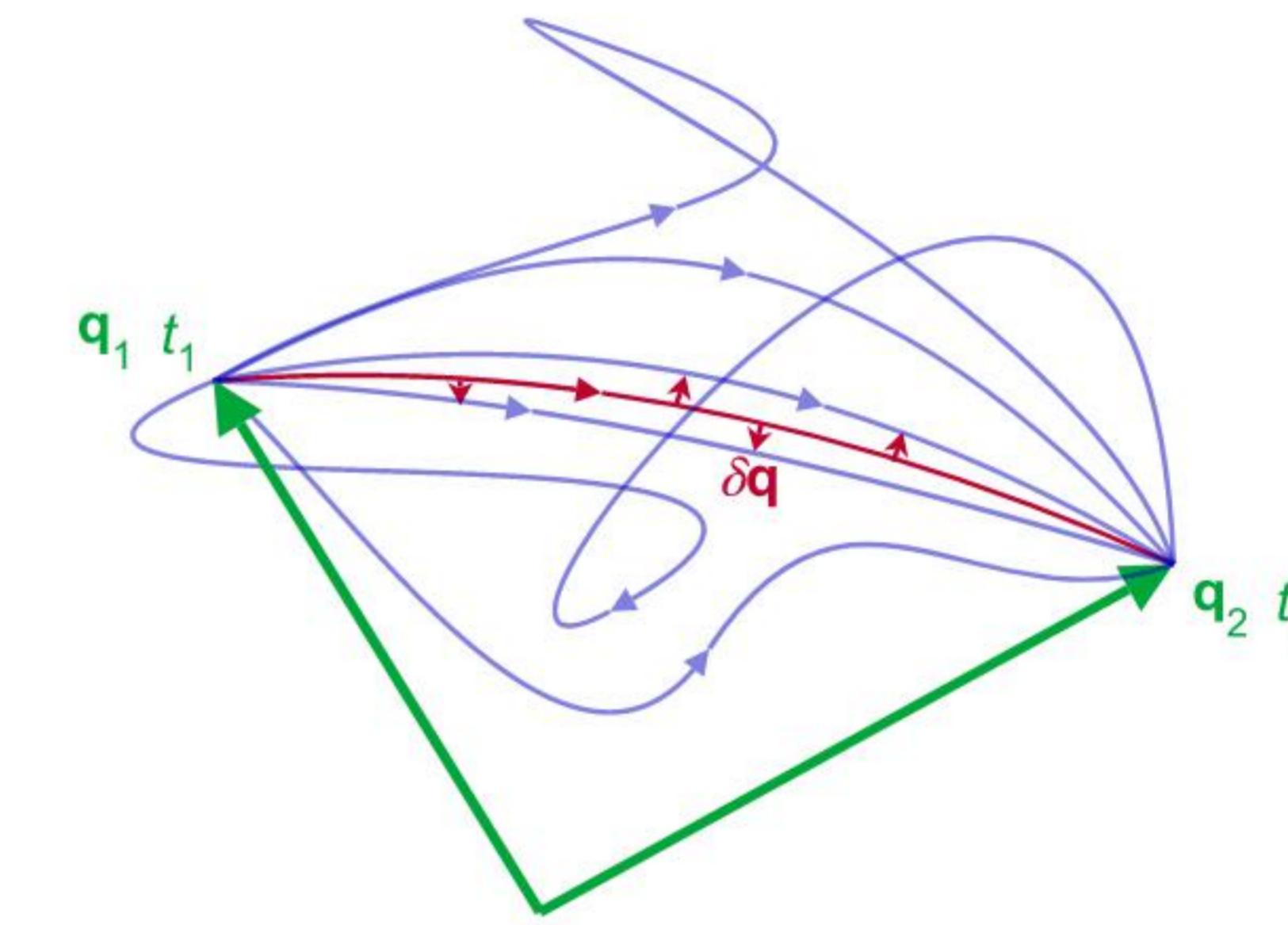
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- This idea of handling constraints can be extended to interconnected rigid bodies (robot arm).

# Lagrangian Function

- Now let  $q \in \mathbb{R}^n$  be the generalized coordinates.
- **Lagrangian function:**  $L(q, \dot{q}) = T(q, \dot{q}) - V(q)$ 
  - $T(q, \dot{q})$ : kinetic energy of system
  - $V(q)$ : potential energy (given by some conservative force, e.g., gravity, electrostatic force)

# The Principle of Stationary Action

- Given a pair of time instants,  $t_1$  and  $t_2$
- What is the curve  $\mathbf{q} : [t_1, t_2] \rightarrow \mathcal{C}$  in the generalized coordinate space  $\mathcal{C}$ ?

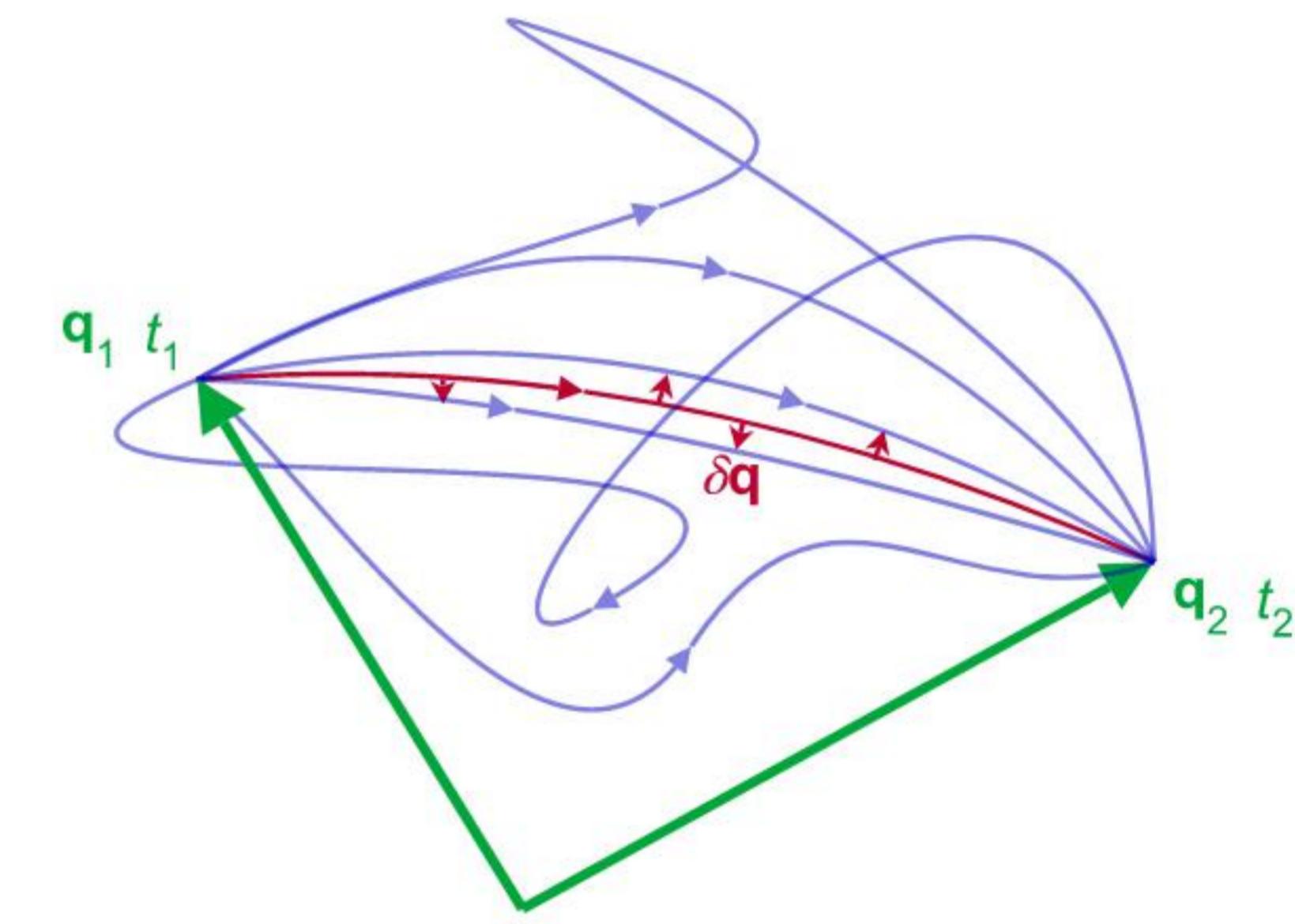


*Wikipedia: Stationary Action Principle*

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- **Action** is defined to be a functional of  $\mathbf{q}(t)$ :

$$S[\mathbf{q}] = \int_{t_1}^{t_2} L(q, \dot{q}) dt = \int_{t_1}^{t_2} [T(\mathbf{q}, \dot{\mathbf{q}}) - V(\mathbf{q})] dt$$



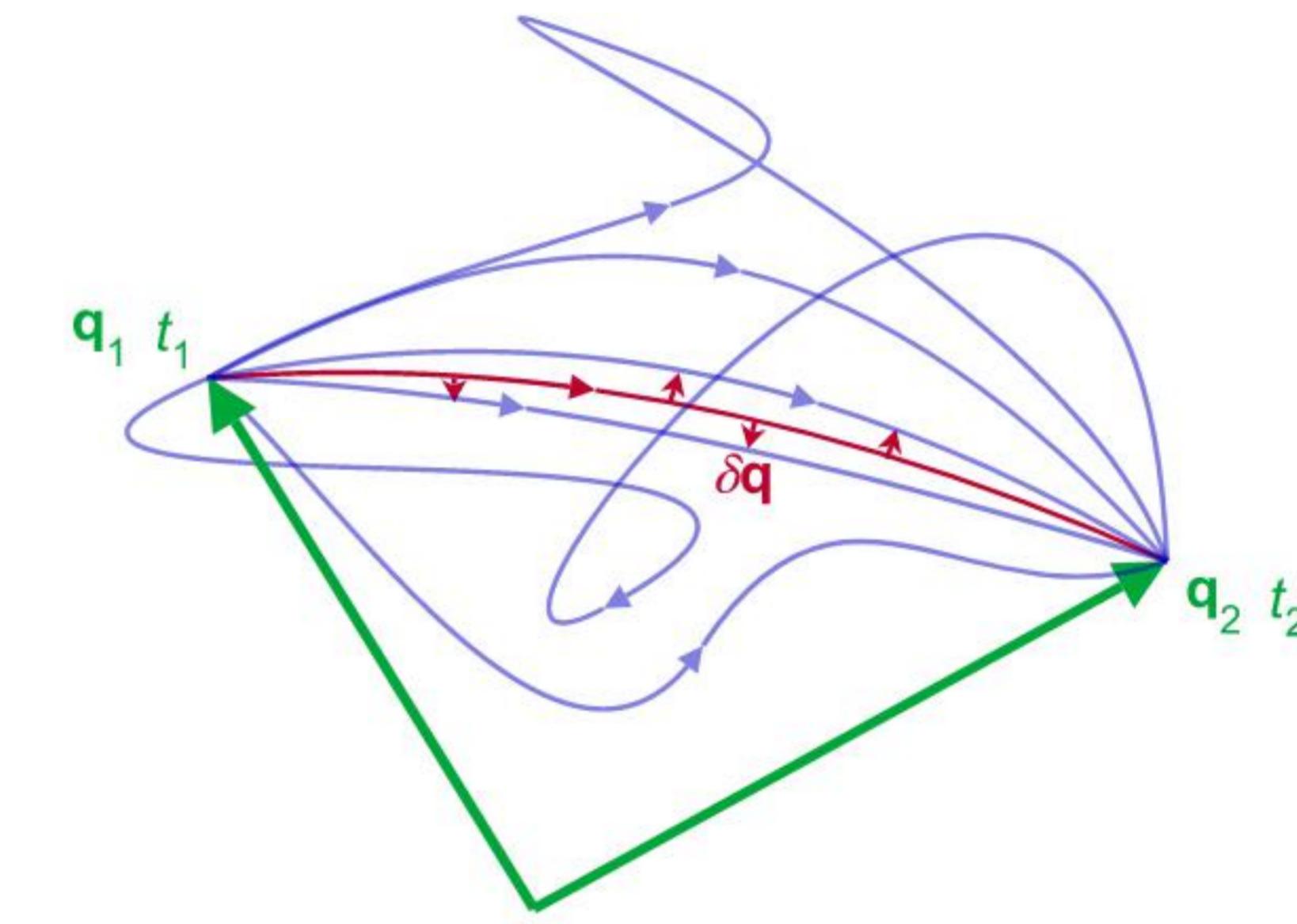
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- The actual curve  $\mathbf{q}(t)$  is a stationary point of the  $S[\mathbf{q}]$ :

$$\forall \boldsymbol{\delta} : [t_1, t_2] \rightarrow \mathcal{C}, \quad \lim_{\epsilon \rightarrow 0} \frac{1}{\epsilon} (\mathcal{S}[\mathbf{q} + \epsilon \boldsymbol{\delta}] - \mathcal{S}[\mathbf{q}]) = \mathbf{0} \quad (1)$$

- Note: Treating  $\mathbf{q}$  as a variable, and (1) is an extension of the first-order optimality condition that we use in calculus:

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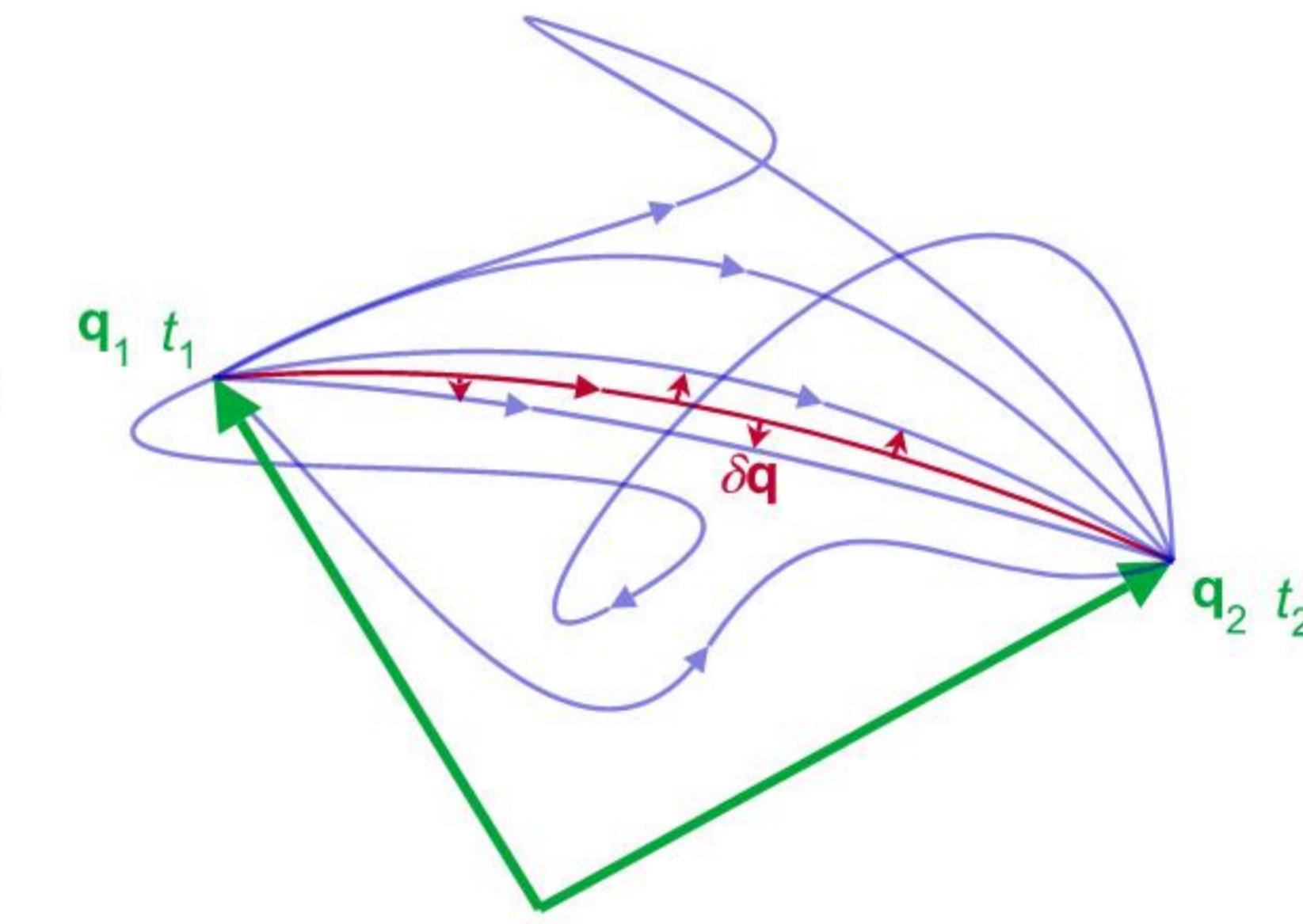
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- Using *variational method*, condition (1) becomes

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q} = 0$$



# A Simple Example

- Consider a point with mass  $m$  and velocity  $v$  is falling down to the ground due to gravity,  $g$  is gravitational acceleration

$$L = \frac{1}{2}mv^2 - mgh$$

- The generalized coordinate is  $h$  and no external force are applied on this system, then

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}} = ma, \frac{\partial L}{\partial q} = mg$$

- Therefore we get

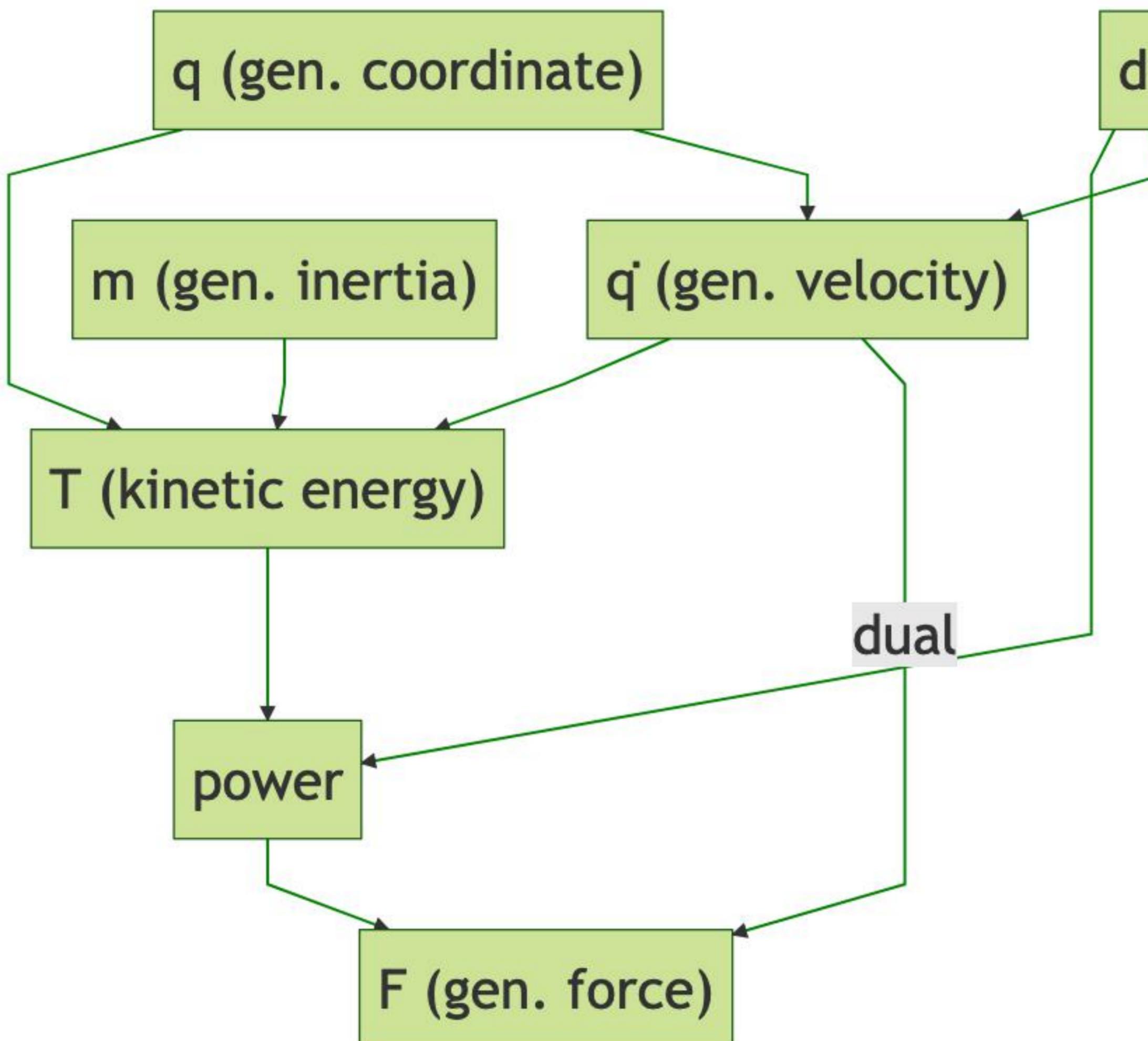
$$ma - mg = 0$$

# Euler-Lagrange Equation

- When there are external non-conservative generalized force  $\mathbf{F} \in \mathbb{R}^n$  added to the system (e.g., torque at robot arm joints), we have the following Euler-Lagrange equation:

$$\mathbf{F} = \frac{d}{dt} \frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q} \quad (\text{Euler-Lagrange Equation})$$

# Logic behind Concepts in Lagrangian Dynamics

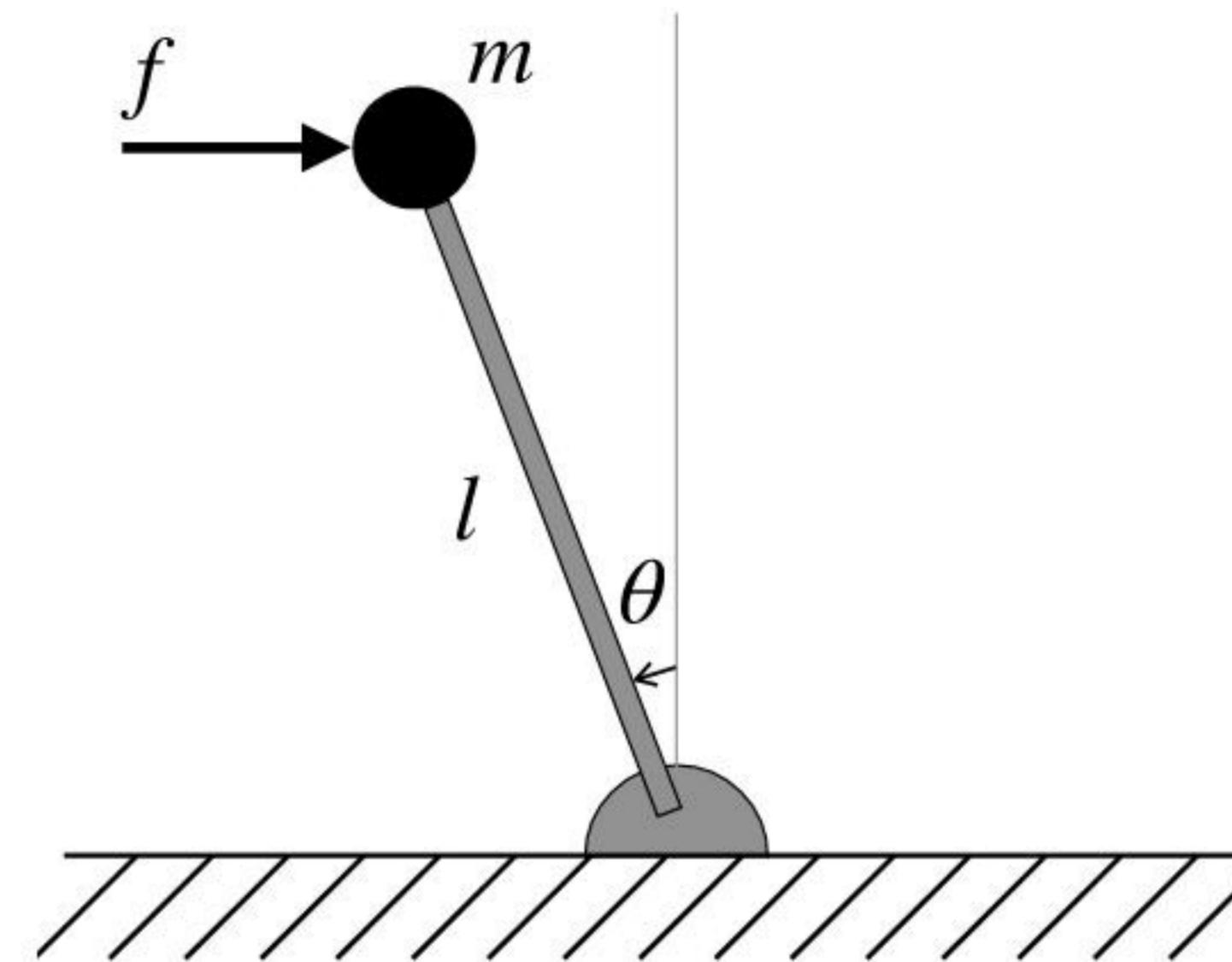


# **Example: Inverted Pendulum**

**(describe using spatial frame)**

# Inverted Pendulum

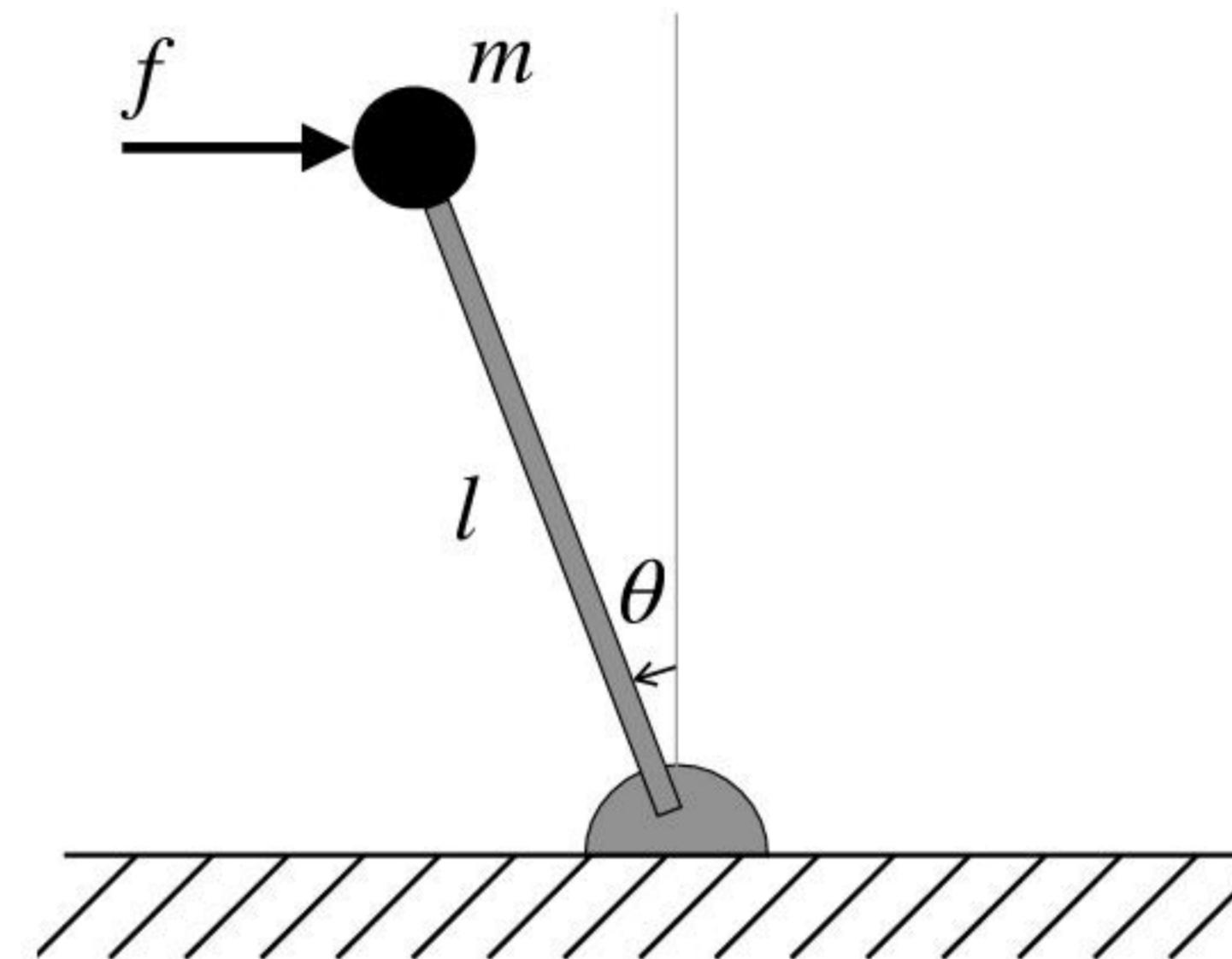
- Kinetic energy:  $T = \frac{1}{2}m(\dot{\theta}l)^2$
- Potential energy:  $V = mgl \cos \theta$



A schematic drawing of the inverted pendulum. The rod is considered massless.

# Generalized Coordinates and Force

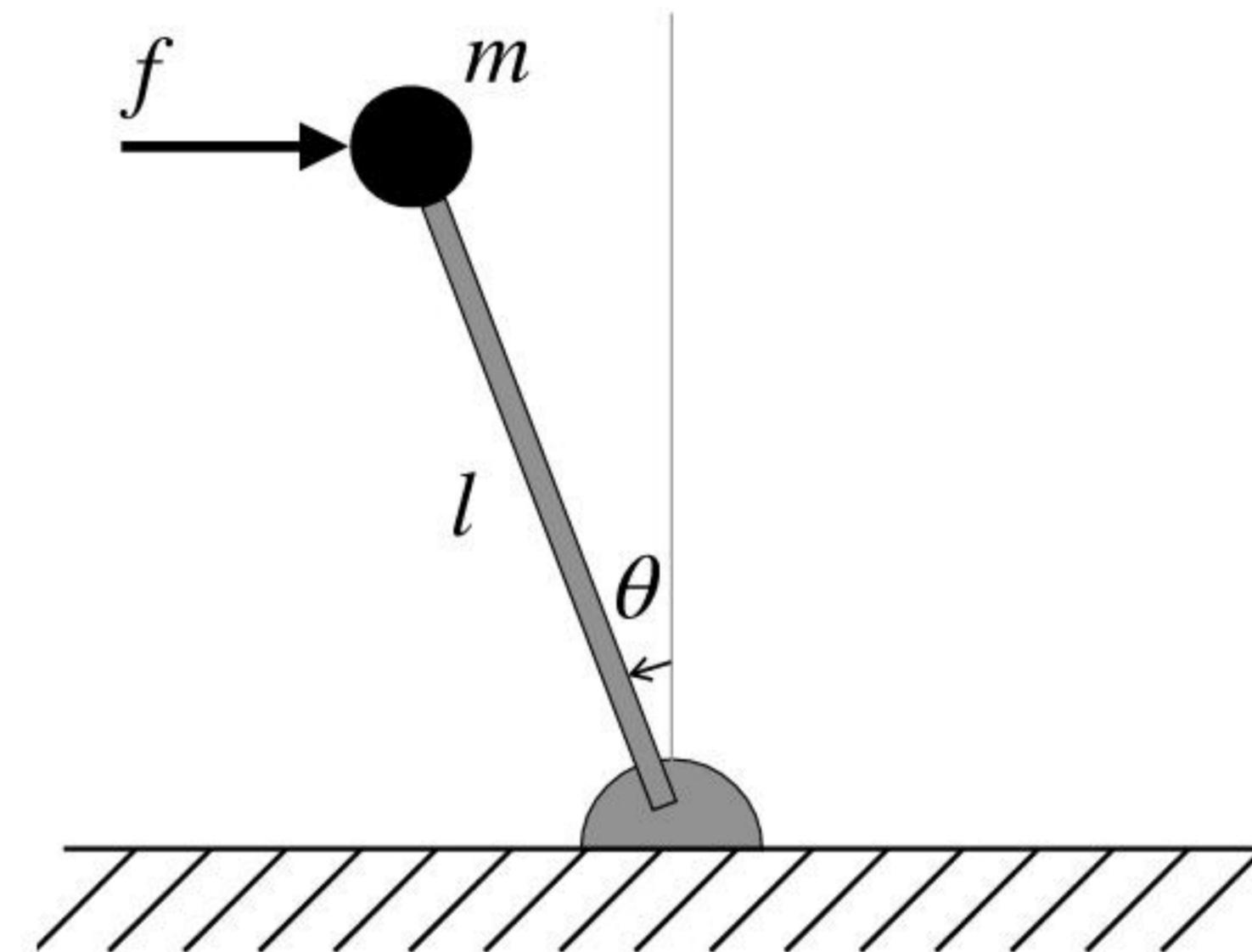
- The generalized coordinate of the system is  $\theta$ .
- What is the generalized force?
- Recall that the inner product of generalized force and generalized velocity is the input power, so we think from the perspective of power



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# Generalized Coordinates and Force

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- What is the generalized force?
- Recall that the inner product of generalized force and generalized velocity is the input power, so we think from the perspective of power
- Assume the coordinate of  $m$  is  $(x, y)$ , so  $P = f \frac{dx}{dt}$
- If  $F$  is a generalized force, then  $F\dot{q} = F \frac{d\theta}{dt} = P = f \frac{dx}{dt}$
- Therefore,  $F = f \frac{dx}{d\theta}$
- But  $x = -l \sin \theta$ , so  $F = -fl \cos \theta$ .



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# Lagrangian Equation

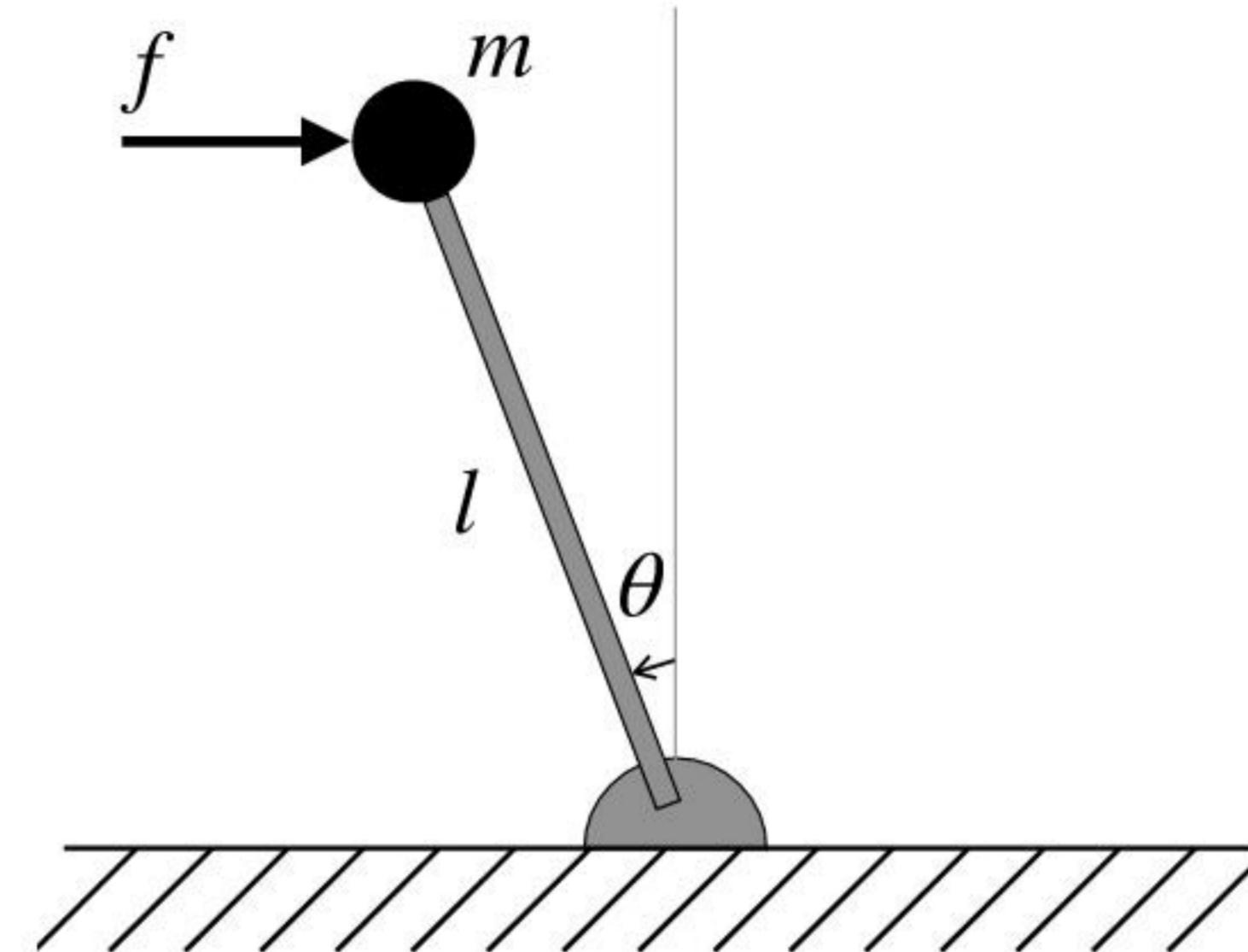
$$L = T - V = \frac{1}{2}ml^2\dot{\theta}^2 - mgl \cos \theta$$

$$F = -fl \cos \theta$$

- Plug in  $F = \frac{d}{dt} \frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q}$ , and we have

$$ml\ddot{\theta} = -f \cos \theta + mg \sin \theta$$

- In Newton's system, the left is  $ma$  and right is the total force tangential to  $\vec{l}$ .



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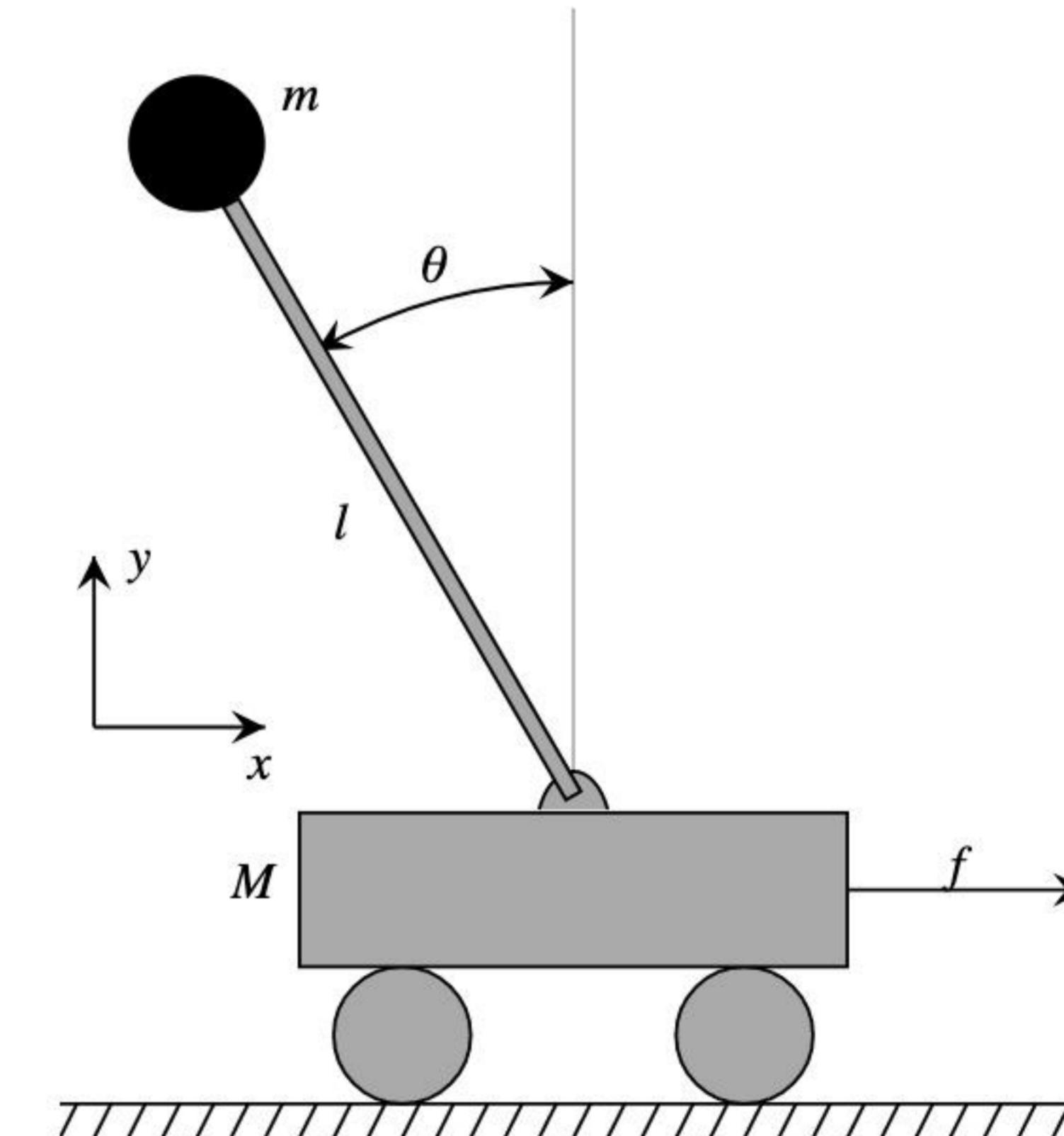
# Example: Cart Pole

# Cart Pole

- Kinetic energy:  $T = \frac{1}{2}Mv_1^2 + \frac{1}{2}mv_2^2$
- Assume the joint position is  $[x(t), 0]^T$ , then
  - $v_1^2 = \dot{x}^2$
  - $v_2^2 = \left(\frac{d}{dt}(x - \ell \sin \theta)\right)^2 + \left(\frac{d}{dt}(\ell \cos \theta)\right)^2$
- Further computation shows that

$$T = \frac{1}{2} (M + m) \dot{x}^2 - m\ell \dot{x} \dot{\theta} \cos \theta + \frac{1}{2} m\ell^2 \dot{\theta}^2$$

- Potential energy:  $V = mgl \cos \theta$

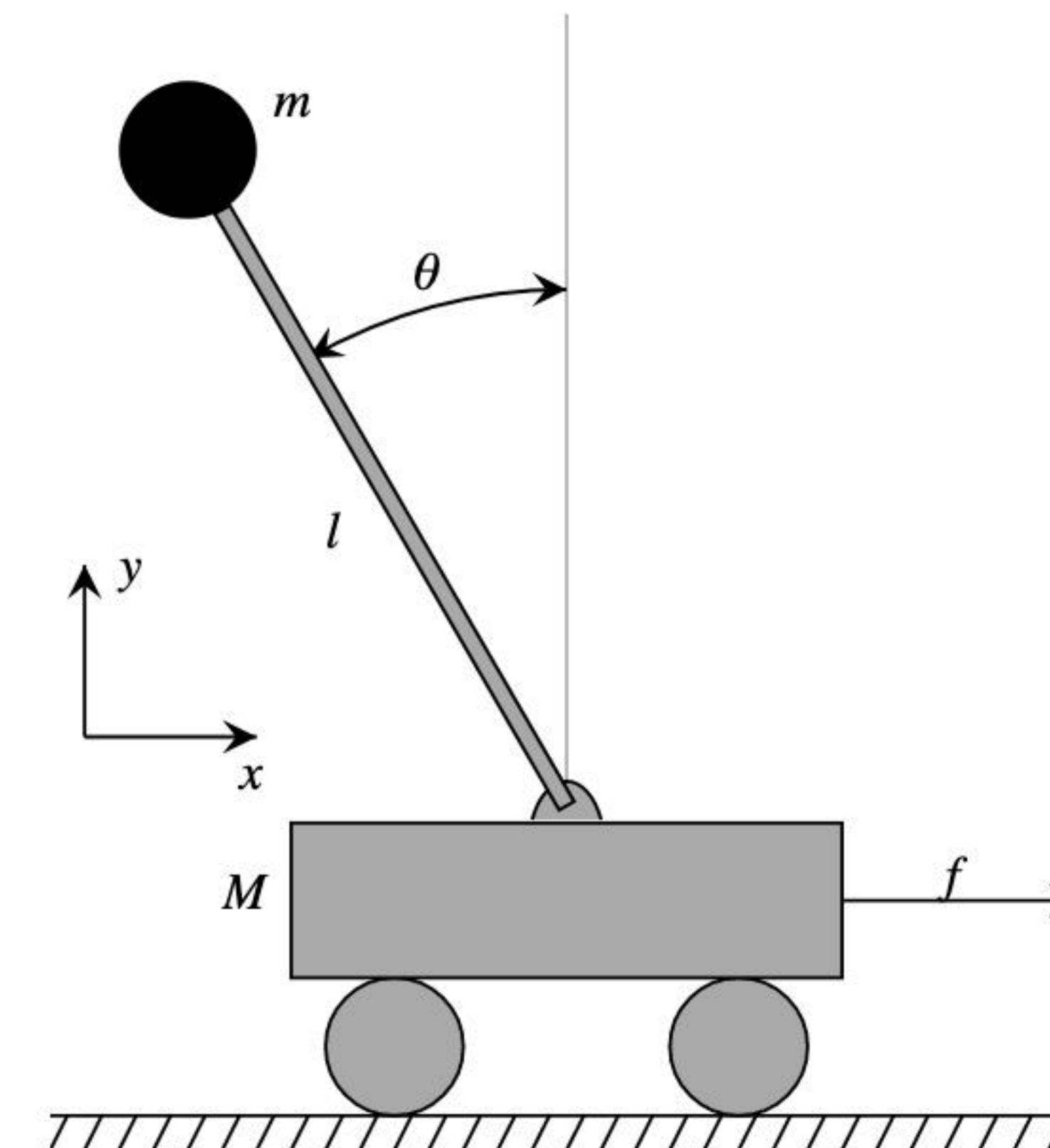


A schematic drawing of the inverted pendulum on a cart. The rod is considered massless.

[https://en.wikipedia.org/wiki/Inverted\\_pendulum](https://en.wikipedia.org/wiki/Inverted_pendulum)

# Generalized Coordinates and Force

- First of all, note that there is an external force  $F$ , and the joint is an Underactuated joint (i.e., *no* torque at the joint)
- The generalized coordinates of the system are  $q = [x, \theta]^T$ , each should have a generalized force.
- What are the generalized forces?

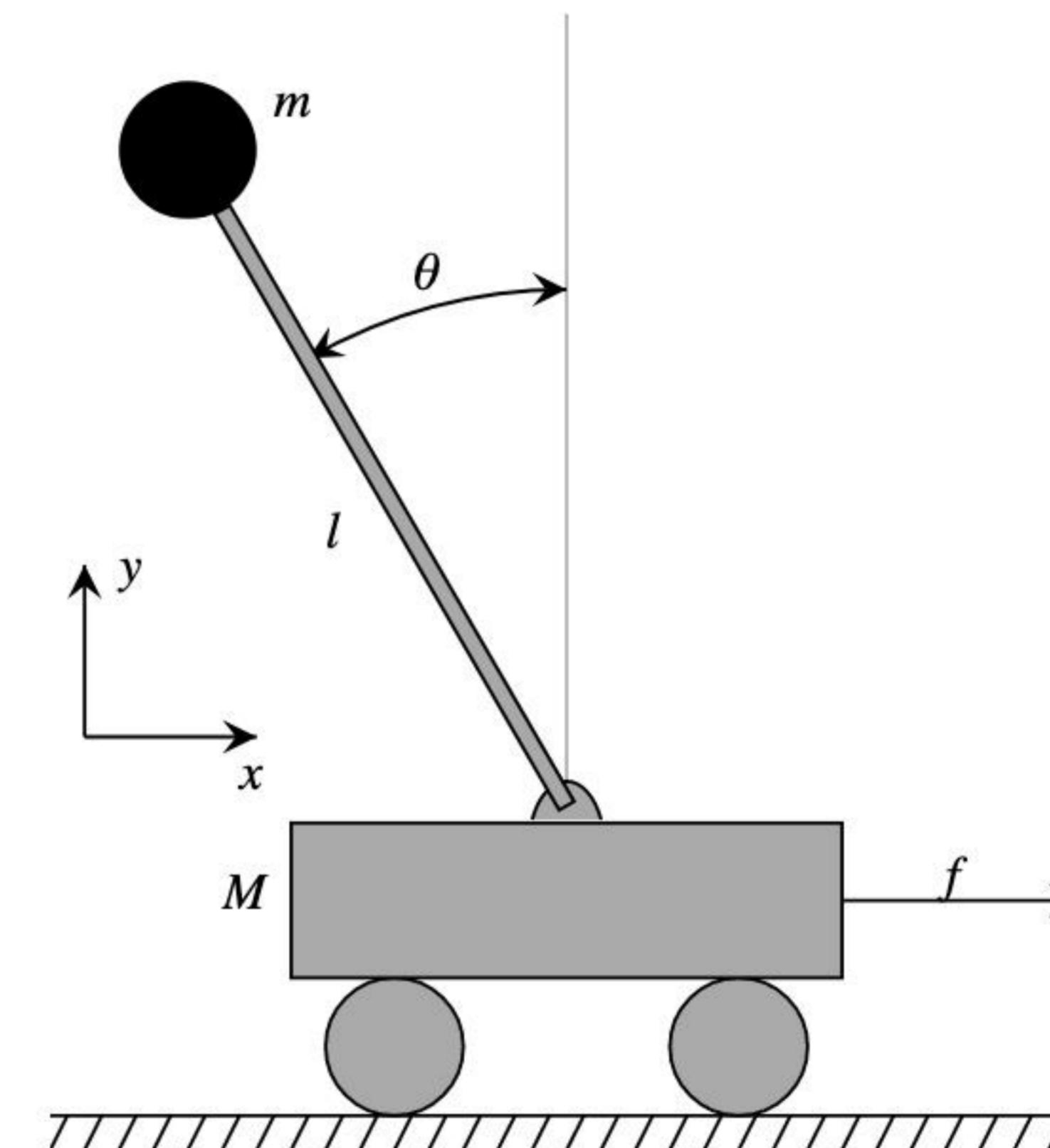


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- What are the generalized forces?
- The input power is  $P = f \frac{dx}{dt}$
- Therefore,  $[f, 0]^T$  is the generalized force dual to  $[x, \theta]^T$



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# Lagrangian Equation

$$L = T - V$$

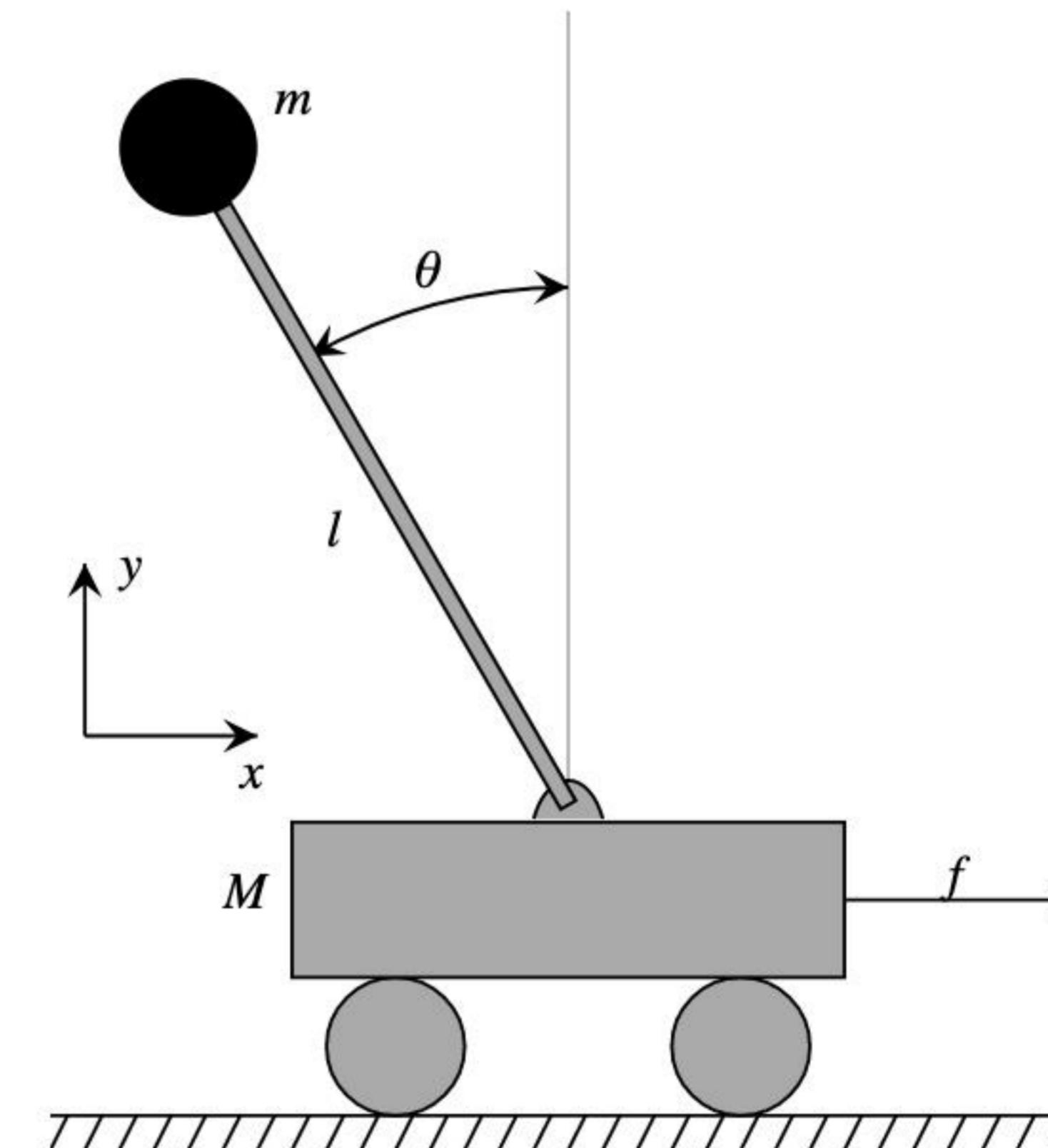
$$= \frac{1}{2} (M + m) \dot{x}^2 - m\ell \dot{x} \dot{\theta} \cos \theta + \frac{1}{2} m\ell^2 \dot{\theta}^2 - mgl \cos \theta$$

$$F = [f, 0]^T$$

- Plug in  $F = \frac{d}{dt} \frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q}$ , and we have

$$(M + m)\ddot{x} - ml \cos \theta \ddot{\theta} + ml \sin \theta \dot{\theta}^2 = f$$

$$l\ddot{\theta} - g \sin \theta - \ddot{x} \cos \theta = 0$$



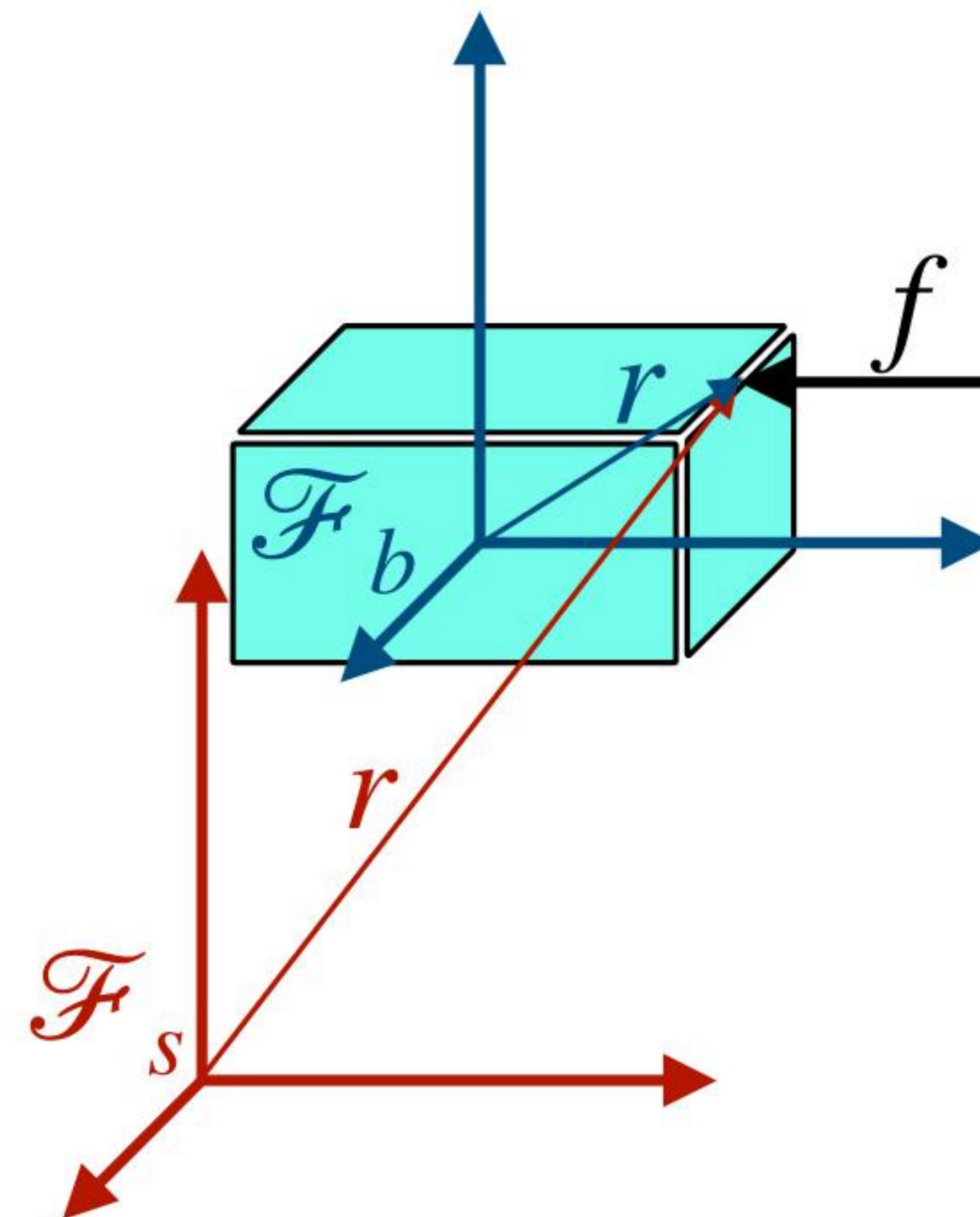
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# **Example: Single-Object Dynamics**

# Setup

- Consider a moving body that is only affected by a force-torque pair (in the sense of the conventional force and torque)



# Prep: Derivative of Acceleration

- Like for velocity, we use the following rule to compute the gradient of velocity in an arbitrary observer's frame:

$$\boldsymbol{a}_{b(t_0)}^{o(t_0)} = \frac{d}{dt} \boldsymbol{v}_{s(t_0) \rightarrow b(t)}^{o(t_0)} \Big|_{t=t_0}$$

- Different from the definition of velocity, acceleration only has one subscript
- We clone a frame  $s(t_0)$  when taking the derivative, so the definition of acceleration is invariant to  $s(t)$

# Prep: Derivative of Acceleration

- Spatial acceleration (spatial frame is an inertia frame):  $\mathbf{a}_{b(t)}^{s(t)} = \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{s(t_0)} \Big|_{t=t_0}$

- Body acceleration:

$$\therefore \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t)} = \frac{d}{dt} R_{b(t) \rightarrow b(t_0)}^{b(t)} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)} = -[\boldsymbol{\omega}^{b(t)}] \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)} + \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)}$$

$$\therefore \boxed{\mathbf{a}_{b(t)}^{b(t)} = \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)} \Big|_{t=t_0} = \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} + [\boldsymbol{\omega}^{b(t_0)}] \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)}}$$

- Using  $\mathbf{v}_{s(t) \rightarrow b(t)}^{s(t)} = R_{s(t) \rightarrow b(t)}^{s(t)} \mathbf{v}_{s(t) \rightarrow b(t)}^{b(t)}$ , you can verify that  $R_{b(t) \rightarrow s(t)}^{b(t)} \mathbf{a}^{s(t)} = \mathbf{a}^{b(t)}$  (so that  $\mathbf{f}^o = m\mathbf{a}^o$  for both the spatial and body frames).
- The second term in the body-frame acceleration is the *Coriolis acceleration*.

# Body-Frame Lagrangian Derivation

- Recall that the kinetic energy  $T$  is:

$$T = \frac{1}{2}(\boldsymbol{\xi}_{s(t) \rightarrow b(t)}^{b(t)})^T \mathfrak{M}^b \boldsymbol{\xi}_{s(t) \rightarrow b(t)}^{b(t)}$$

where  $\mathfrak{M}^b = \begin{bmatrix} m\text{Id}_{3 \times 3} & 0 \\ 0 & \mathbf{I}^b \end{bmatrix} \in \mathbb{R}^{6 \times 6}$  and  $\boldsymbol{\xi}^b = \begin{bmatrix} \mathbf{v}^b \\ \boldsymbol{\omega}^b \end{bmatrix}$

# Generalized Velocity and Force

- Recall that we introduced  $\mathbf{F}^{b(t)} = \begin{bmatrix} \mathbf{f}^{b(t)} \\ \boldsymbol{\tau}^{b(t)} \end{bmatrix}$  and  $(\mathbf{F}^{b(t)})^T \boldsymbol{\xi}^{b(t)} = \frac{dT}{dt}$ 
  - implies that  $(\boldsymbol{\xi}^{b(t)}, \mathbf{F}^{b(t)})$  is a dual pair
  - $\mathbf{F}^{b(t)}$  is a generalized force and  $\boldsymbol{\xi}^{b(t)}$  is a generalized velocity

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  - $\mathbf{F}^{b(t)}$  is a generalized force and  $\boldsymbol{\xi}^{b(t)}$  is a generalized velocity
- Therefore, we can plug them in Euler-Lagrange equation:

$$F = \frac{d}{dt} \frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q} \quad (\text{Euler-Lagrange Equation})$$

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- Recall that we have derived that

$$L = \frac{1}{2} (\boldsymbol{\xi}^{b(t)})^T \mathfrak{M}^{b(t)} \boldsymbol{\xi}^{b(t)} = \frac{1}{2} m \|\mathbf{v}^{b(t)}\|^2 + \frac{1}{2} (\boldsymbol{\omega}^{b(t)})^T \mathbf{I}^b \boldsymbol{\omega}^{b(t)}$$

- Therefore,  $\frac{\partial L}{\partial \mathbf{v}^{b(t)}} = m \mathbf{v}^{b(t)}$ ,  $\frac{\partial L}{\partial \boldsymbol{\omega}^{b(t)}} = \mathbf{I}^b \boldsymbol{\omega}^{b(t)}$

# Body-Frame Lagrangian Derivation

$$\frac{d}{dt} \frac{\partial L}{\partial \xi^{b(t)}} = \mathbf{F}^{b(t)}, \quad \frac{\partial L}{\partial \mathbf{v}^{b(t)}} = m\mathbf{v}^{b(t)}, \quad \frac{\partial L}{\partial \boldsymbol{\omega}^{b(t)}} = \mathbf{I}^b \boldsymbol{\omega}^{b(t)}$$

- Therefore,

$$\mathbf{f}^{b(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \mathbf{v}^{b(t_0)}} = m \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)} = m \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} + m [\boldsymbol{\omega}_{s(t_0) \rightarrow b(t_0)}^{b(t_0)}] \mathbf{v}_{s(t_0) \rightarrow b(t_0)}^{b(t_0)}$$

$$\boldsymbol{\tau}^{b(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \boldsymbol{\omega}^{b(t_0)}} = \frac{d}{dt} \mathbf{I}^b \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} = \mathbf{I}^b \frac{d}{dt} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} + [\boldsymbol{\omega}^{b(t_0)}] \mathbf{I}^b \boldsymbol{\omega}^{b(t_0)}$$

# Body-Frame Lagrangian Derivation

$$\frac{d}{dt} \frac{\partial L}{\partial \xi^{b(t)}} = \mathbf{F}^{b(t)}, \quad \frac{\partial L}{\partial \mathbf{v}^{b(t)}} = m\mathbf{v}^{b(t)}, \quad \frac{\partial L}{\partial \boldsymbol{\omega}^{b(t)}} = \mathbf{I}^b \boldsymbol{\omega}^{b(t)}$$

- Therefore,

$$\mathbf{f}^{b(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \mathbf{v}^{b(t_0)}} = m \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t_0)} = m \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} + m [\boldsymbol{\omega}_{s(t_0) \rightarrow b(t_0)}^{b(t_0)}] \mathbf{v}_{s(t_0) \rightarrow b(t_0)}^{b(t_0)}$$

$$\boldsymbol{\tau}^{b(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \boldsymbol{\omega}^{b(t_0)}} = \frac{d}{dt} \mathbf{I}^b \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} = \mathbf{I}^b \frac{d}{dt} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{b(t)} \Big|_{t=t_0} + [\boldsymbol{\omega}^{b(t_0)}] \mathbf{I}^b \boldsymbol{\omega}^{b(t_0)}$$

- In matrix form, we have the famous **body-frame Newton-Euler equation**:

$$\begin{bmatrix} m\text{Id}_{3 \times 3} & 0 \\ 0 & \mathbf{I}^b \end{bmatrix} \begin{bmatrix} \dot{\mathbf{v}}^b \\ \dot{\boldsymbol{\omega}}^b \end{bmatrix} + \begin{bmatrix} \boldsymbol{\omega}^b \times m\mathbf{v}^b \\ \boldsymbol{\omega}^b \times \mathbf{I}^b \boldsymbol{\omega}^b \end{bmatrix} = \begin{bmatrix} \mathbf{f}^b \\ \boldsymbol{\tau}^b \end{bmatrix}$$

in which the precise definition of symbols are as above.

# Spatial-Frame Lagrangian Derivation

- The kinetic energy  $T$  can also be computed by spatial frame velocities:

$$T = \frac{1}{2}m\|\boldsymbol{v}^{s(t)}\|^2 + \frac{1}{2}(\boldsymbol{\omega}^{s(t)})^T \boldsymbol{I}^s \boldsymbol{\omega}^{s(t)}$$

- Force-velocity duality:  $\begin{bmatrix} \boldsymbol{f}^{s(t)} \\ \boldsymbol{\tau}^{s(t)} \end{bmatrix}$  and  $\begin{bmatrix} \boldsymbol{v}^{s(t)} \\ \boldsymbol{\omega}^{s(t)} \end{bmatrix}$  form a duality pair

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- Justification:
  - The translational kinetic energy can be justified by  $\boldsymbol{v}^{s(t)} = R_{s(t) \rightarrow b(t)}^{s(t)} \boldsymbol{v}^{b(t)}$
  - The rotational kinetic energy can be justified by  $\boldsymbol{\omega}^{s(t)} = R_{s(t) \rightarrow b(t)}^{s(t)} \boldsymbol{\omega}^{b(t)}$  and  
 $\boldsymbol{I}^{s(t)} = R_{s(t) \rightarrow b(t)}^{s(t)} \boldsymbol{I}^b (R_{s(t) \rightarrow b(t)}^{s(t)})^T$
  - The force-velocity duality can be justified by  $\boldsymbol{f}^{s(t)} = R_{s(t) \rightarrow b(t)}^{s(t)} \boldsymbol{f}^{b(t)}$ ,  $\boldsymbol{\tau}^{s(t)} = R_{s(t) \rightarrow b(t)}^{s(t)} \boldsymbol{\tau}^{b(t)}$ , and the frame rules for  $\boldsymbol{v}$  and  $\boldsymbol{\omega}$

# Spatial-Frame Lagrangian Derivation

$$\frac{\partial L}{\partial \mathbf{v}^{s(t)}} = m \mathbf{v}^{s(t)}, \quad \frac{\partial L}{\partial \boldsymbol{\omega}^{s(t)}} = \mathbf{I}^b \boldsymbol{\omega}^{s(t)}$$

- Therefore,

$$\mathbf{f}^{s(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \mathbf{v}^{s(t_0)}} = m \frac{d}{dt} \mathbf{v}^{s(t_0)} = m \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{s(t_0)} \Big|_{t=t_0}$$

$$\boldsymbol{\tau}^{s(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \boldsymbol{\omega}^{s(t_0)}} = \frac{d}{dt} \mathbf{I}^{s(t)} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{s(t_0)} \Big|_{t=t_0} = [\boldsymbol{\omega}_{s(t_0) \rightarrow b(t_0)}^{s(t_0)}] \mathbf{I}^{s(t_0)} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t_0)}^{s(t_0)} + \mathbf{I}^{s_0} \frac{d}{dt} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{s(t)} \Big|_{t=t_0}$$

# Spatial-Frame Lagrangian Derivation

$$\frac{\partial L}{\partial \mathbf{v}^{s(t)}} = m\mathbf{v}^{s(t)}, \quad \frac{\partial L}{\partial \boldsymbol{\omega}^{s(t)}} = \mathbf{I}^b \boldsymbol{\omega}^{s(t)}$$

- Therefore,

$$\mathbf{f}^{s(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \mathbf{v}^{s(t_0)}} = m \frac{d}{dt} \mathbf{v}^{s(t_0)} = m \frac{d}{dt} \mathbf{v}_{s(t_0) \rightarrow b(t)}^{s(t_0)} \Big|_{t=t_0}$$

$$\boldsymbol{\tau}^{s(t_0)} = \frac{d}{dt} \frac{\partial L}{\partial \boldsymbol{\omega}^{s(t_0)}} = \frac{d}{dt} \mathbf{I}^{s(t)} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{s(t_0)} \Big|_{t=t_0} = [\boldsymbol{\omega}_{s(t_0) \rightarrow b(t_0)}^{s(t_0)}] \mathbf{I}^{s(t_0)} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t_0)}^{s(t_0)} + \mathbf{I}^{s_0} \frac{d}{dt} \boldsymbol{\omega}_{s(t_0) \rightarrow b(t)}^{s(t)} \Big|_{t=t_0}$$

- In matrix form, we have the famous **spatial frame Newton-Euler equation**:

$$\begin{bmatrix} m\text{Id}_{3 \times 3} & 0 \\ 0 & \mathbf{I}^s \end{bmatrix} \begin{bmatrix} \dot{\mathbf{v}}^s \\ \dot{\boldsymbol{\omega}}^s \end{bmatrix} + \begin{bmatrix} \mathbf{0} \\ \boldsymbol{\omega}^s \times \mathbf{I}^s \boldsymbol{\omega}^s \end{bmatrix} = \begin{bmatrix} \mathbf{f}^s \\ \boldsymbol{\tau}^s \end{bmatrix}$$

in which the precise definition of symbols are as above.

# **Example: Robot Arm**

# Robot Arm

- For kinematic chains with  $n$  joints, it is convenient and always possible to choose the joint angles  $\theta = (\theta_1, \dots, \theta_n)$  and the joint torques  $\tau = (\tau_1, \dots, \tau_n)$  as the generalized coordinates and generalized forces, respectively.
  - If joint  $i$  is revolute:  $\theta_i$  joint angle and  $\tau_i$  is joint torque
  - If joint  $i$  is prismatic:  $\theta_i$  joint position and  $\tau_i$  is joint force
- Lagrangian Equations:

$$\tau_i = \frac{d}{dt} \frac{\partial L}{\partial \dot{\theta}_i} - \frac{\partial L}{\partial \theta_i}$$

# Some Notations

For each link  $i = 1, \dots, n$ ,  $\mathcal{F}_i$  is attached to the center of mass of link  $i$ . All the following quantities are expressed in  $\mathcal{F}_i$

- $\boldsymbol{\xi}_i^b$ : twist of link  $i$
- $m_i$ : mass;  $\mathbf{I}_i^b$ : rotational inertia matrix;
- $\mathfrak{M}_i^b = \begin{bmatrix} m_i \text{Id}_{3 \times 3} & 0 \\ 0 & \mathbf{I}_i^b \end{bmatrix}$ : body inertia matrix
- Kinetic energy of link  $i$ :  $T_i = \frac{1}{2} (\boldsymbol{\xi}_i^b)^T \mathfrak{M}_i^b \boldsymbol{\xi}_i^b$
- $J_i^b \in \mathbb{R}^{6 \times n}$ : body Jacobian of link  $i$

# Kinetic and Potential Energies

- Total kinetic energy:

$$T(\theta, \dot{\theta}) = \frac{1}{2} \sum_{i=1}^n (\boldsymbol{\xi}_i^b)^T \mathfrak{M}_i^b \boldsymbol{\xi}_i^b = \frac{1}{2} \dot{\theta}^T \underbrace{\left( \sum_{i=1}^n (J_i^b(\theta) \mathfrak{M}_i^b J_i^b(\theta)) \right)}_{\mathbf{M}^b(\theta)} \dot{\theta} := \frac{1}{2} \dot{\theta}^T \mathbf{M}^b(\theta) \dot{\theta}$$

- Potential energy:

$$V(\theta) = \sum_{i=1}^n m_i g h_i(\theta)$$

- $h_i(\theta)$ : height of center of mass of link  $i$

# Lagrangian Equation

- Plug  $L = T - V$  into  $F = \frac{d}{dt} \frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q}$ , and we have
- $\tau_i = \frac{d}{dt} \frac{\partial L}{\partial \dot{\theta}_i} - \frac{\partial L}{\partial \theta_i}$

$$\tau_i = \sum_{j=1}^n M_{ij}^b(\theta) \ddot{\theta}_j + \sum_{j=1}^n \sum_{k=1}^n \Gamma_{ijk}^b(\theta) \dot{\theta}_j \dot{\theta}_k + \frac{\partial V}{\partial \theta_i}$$

$M_{ij}^b$  is the  $(i, j)$ -th entry of matrix  $\mathbf{M}^b$

- $\Gamma_{ijk}^b(\theta)$  is called the **Christoffel symbols of the first kind**

$$\Gamma_{ijk}^b(\theta) = \frac{1}{2} \left( \frac{\partial M_{ij}^b}{\partial \theta_k} + \frac{\partial M_{ik}^b}{\partial \theta_j} - \frac{\partial M_{jk}^b}{\partial \theta_i} \right)$$

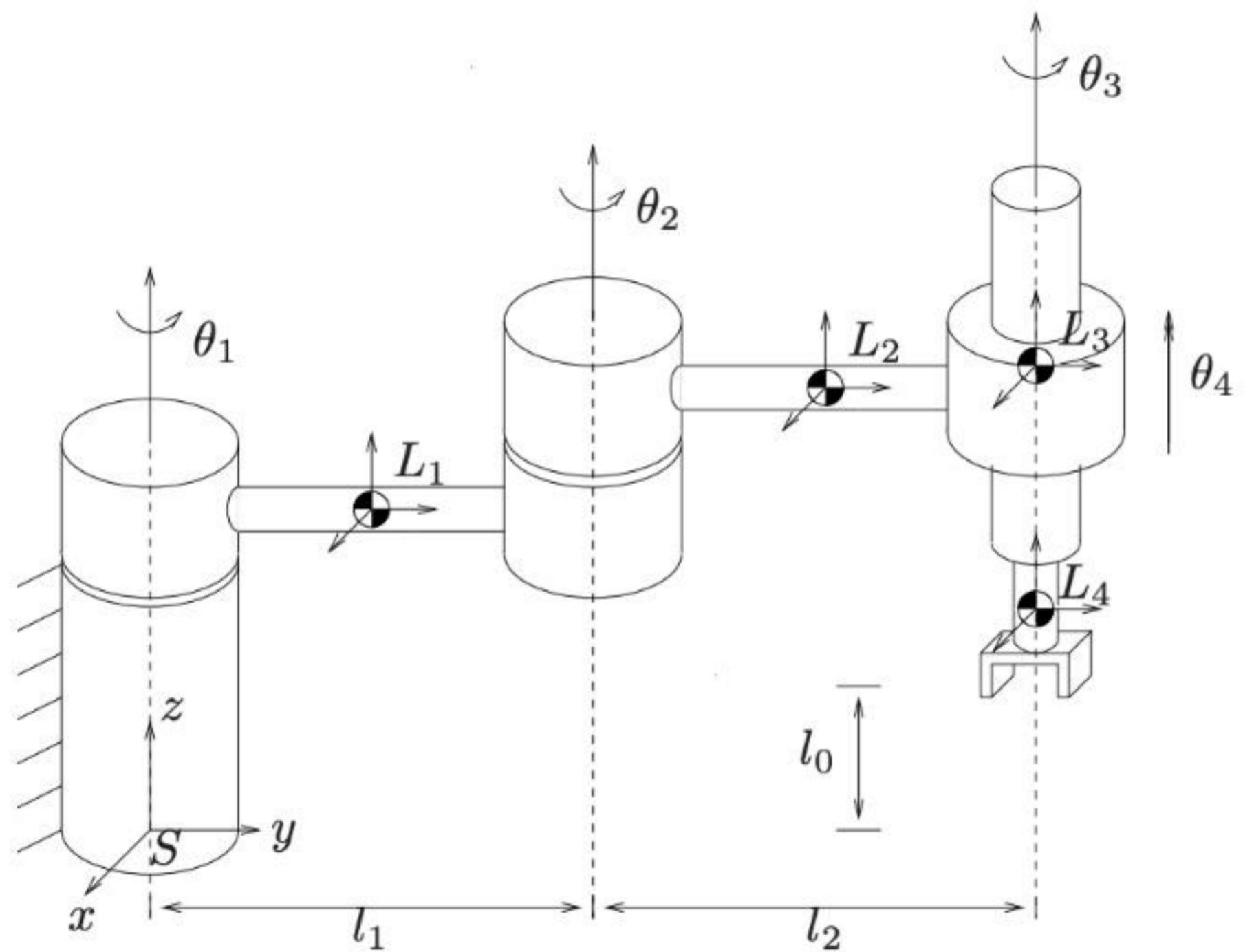
# Lagrangian Equation

- Lagrangian equation in vector form:

$$\tau = \mathbf{M}^b(\theta)\ddot{\theta} + C^b(\theta, \dot{\theta})\dot{\theta} + g^b(\theta)$$

- $C_{ij}^b(\theta, \dot{\theta}) := \sum_{k=1}^n \Gamma_{ijk}^b \dot{\theta}_k$  is called the **Coriolis matrix**
  - Recall that in the body-frame Newton Euler equation, we also have a Coriolis term that comes from the derivative of rotational inertia. It was used to compensate for the rotational acceleration of the body frame
  - This  $C_{ij}^b(\theta, \dot{\theta})$  also comes from taking the derivative of  $\mathbf{M}^b$  w.r.t.  $\theta$ . Because  $\mathbf{M}^b$  and  $\boldsymbol{\xi}^b$  are described in the body frame in our derivation, we also need this Coriolis term to compensate for the movement of the body frame.
- $g^b(\theta)$  is due to gravity in our derivation. If there are other external forces (e.g., friction), it would also show up here.

- Equations for a simple arm



$$M_{11} = I_{y2}s_2^2 + I_{y3}s_{23}^2 + I_{z1} + I_{z2}c_2^2 + I_{z3}c_{23}^2 + m_2r_1^2c_2^2 + m_3(l_1c_2 + r_2c_{23})^2$$

$$M_{12} = 0$$

$$M_{13} = 0$$

$$M_{21} = 0$$

$$M_{22} = I_{x2} + I_{x3} + m_3l_1^2 + m_2r_1^2 + m_3r_2^2 + 2m_3l_1r_2c_3$$

$$M_{23} = I_{x3} + m_3r_2^2 + m_3l_1r_2c_3$$

$$M_{31} = 0$$

$$M_{32} = I_{x3} + m_3r_2^2 + m_3l_1r_2c_3$$

$$M_{33} = I_{x3} + m_3r_2^2.$$

$$\Gamma_{112} = (I_{y2} - I_{z2} - m_2r_1^2)c_2s_2 + (I_{y3} - I_{z3})c_{23}s_{23} - m_3(l_1c_2 + r_2c_{23})(l_1s_2 + r_2s_{23})$$

$$\Gamma_{113} = (I_{y3} - I_{z3})c_{23}s_{23} - m_3r_2s_{23}(l_1c_2 + r_2c_{23})$$

$$\Gamma_{121} = (I_{y2} - I_{z2} - m_2r_1^2)c_2s_2 + (I_{y3} - I_{z3})c_{23}s_{23} - m_3(l_1c_2 + r_2c_{23})(l_1s_2 + r_2s_{23})$$

$$\Gamma_{131} = (I_{y3} - I_{z3})c_{23}s_{23} - m_3r_2s_{23}(l_1c_2 + r_2c_{23})$$

$$\Gamma_{211} = (I_{z2} - I_{y2} + m_2r_1^2)c_2s_2 + (I_{z3} - I_{y3})c_{23}s_{23} + m_3(l_1c_2 + r_2c_{23})(l_1s_2 + r_2s_{23})$$

$$\Gamma_{223} = -l_1m_3r_2s_3$$

$$\Gamma_{232} = -l_1m_3r_2s_3$$

$$\Gamma_{233} = -l_1m_3r_2s_3$$

$$\Gamma_{311} = (I_{z3} - I_{y3})c_{23}s_{23} + m_3r_2s_{23}(l_1c_2 + r_2c_{23})$$

$$\Gamma_{322} = l_1m_3r_2s_3$$

$$\begin{bmatrix} 0 \\ -(m_2gr_1 + m_3gl_1)\cos\theta_2 - m_3r_2\cos(\theta_2 + \theta_3) \\ -m_3gr_2\cos(\theta_2 + \theta_3) \end{bmatrix}$$

## • Equations for PUMA 560 Arm

$$\begin{aligned}
I_2 &= I_{xx2} + m_2 * (r_{x2}^2 + r_{y2}^2) + (m_3 + m_4 + m_5 + m_6) * a_2^2 ; \\
I_3 &= -I_{xz2} + I_{yz2} + (m_3 + m_4 + m_5 + m_6) * a_2^2 \\
&\quad m_2 * r_{x2}^2 - m_2 * r_{y2}^2 ; \\
I_4 &= m_2 * r_{x2} * (d_2 + r_{z2}) + m_5 * a_2 * r_{z2} \\
&\quad + (m_3 + m_4 + m_5 + m_6) * a_2 * (d_2 + d_3) ; \\
I_5 &= -m_5 * a_2 * r_{x3} + (m_4 + m_5 + m_6) * a_2 * d_4 + m_4 * a_2 * r_{z4} ; \\
I_6 &= I_{zz5} + m_5 * r_{z5}^2 + m_4 * a_2^2 + m_4 * (d_4 + r_{z4})^2 + I_{yy4} \\
&\quad + m_5 * a_3^2 + m_5 * d_4^2 + I_{zz5} + m_6 * a_2^2 + m_6 * d_4 \\
&\quad + m_6 * r_{z6}^2 + I_{zz6} ; \\
I_7 &= m_5 * r_{x2}^2 + I_{zz5} - I_{yy3} + m_4 * r_{z4}^2 + 2 * m_4 * d_4 * r_{z4} \\
&\quad + (m_4 + m_5 + m_6) * (d_4^2 - a_3^2) + I_{yy3} - I_{zz4} + I_{zz5} \\
&\quad - I_{yy5} + m_6 * r_{z6}^2 - I_{zz5} + I_{zz6} ; \\
I_8 &= -m_4 * (d_2 + d_3) * (d_4 + r_{z4}) - (m_5 + m_6) * (d_2 + d_3) * d_4 \\
&\quad m_5 * r_{x2} * r_{z5} + m_5 * (d_2 + d_3) * r_{y5} ; \\
I_9 &= m_2 * r_{y2} * (d_2 + r_{z2}) ; \\
I_{10} &= 2 * m_4 * a_2 * r_{z4} + 2 * (m_4 + m_5 + m_6) * a_3 * d_4 ; \\
I_{11} &= -2 * m_5 * r_{z2} * r_{y2} ; \\
I_{12} &= (m_4 + m_5 + m_6) * a_2 * a_3 ; \\
I_{13} &= (m_4 + m_5 + m_6) * a_3 * (d_2 + d_3) ; \\
I_{14} &= I_{zz4} + I_{yy5} + I_{zz6} ; \\
I_{15} &= m_6 * d_4 * r_{z6} ; \\
I_{16} &= m_6 * a_2 * r_{z6} ; \\
I_{17} &= I_{zz5} + I_{zz6} + m_6 * r_{z6}^2 ; \\
I_{18} &= m_6 * (d_2 + d_3) * r_{z6} ; \\
I_{19} &= I_{yy5} - I_{zz4} + I_{zz5} - I_{yy5} + m_6 * r_{x2}^2 + I_{zz6} - I_{zz5} ; \\
I_{20} &= I_{yy5} - I_{zz5} - m_6 * r_{z6}^2 + I_{zz6} - I_{zz5} ; \\
I_{21} &= I_{zz4} - I_{yy4} + I_{zz5} - I_{zz5} ; \\
I_{22} &= m_6 * a_3 * r_{z6} ; \\
I_{23} &= I_{zz6} ;
\end{aligned}$$

**Part II. Gravitational Constants**

$$\begin{aligned}
g_1 &= -g * ((m_3 + m_4 + m_5 + m_6) * a_2 + m_2 * r_{z2}) ; \\
g_2 &= g * (m_3 * r_{y5} - (m_4 + m_5 + m_6) * d_4 - m_4 * r_{z4}) ; \\
g_3 &= g * m_2 * r_{y2} ; \\
g_4 &= -g * (m_4 + m_5 + m_6) * a_3 ; \\
g_5 &= -g * m_6 * r_{z6} ;
\end{aligned}$$

**Table A3.** Computed Values for the Constants Appearing in the Equations of Forces of Motion.  
(Inertial constants have units of kilogram meters-squared)

$$\begin{aligned}
I_1 &= 1.43 \pm 0.05 & I_2 &= 1.75 \pm 0.07 \\
I_3 &= 1.38 \pm 0.05 & I_4 &= 6.90 \times 10^{-1} \pm 0.20 \times 10^{-1} \\
I_5 &= 3.72 \times 10^{-1} \pm 0.31 \times 10^{-1} & I_6 &= 3.33 \times 10^{-1} \pm 0.16 \times 10^{-1} \\
I_7 &= 2.98 \times 10^{-1} \pm 0.29 \times 10^{-1} & I_8 &= -1.34 \times 10^{-1} \pm 0.14 \times 10^{-1} \\
I_9 &= 2.38 \times 10^{-2} \pm 1.20 \times 10^{-2} & I_{10} &= -2.13 \times 10^{-2} \pm 0.22 \times 10^{-2} \\
I_{11} &= -1.42 \times 10^{-2} \pm 0.70 \times 10^{-2} & I_{12} &= -1.10 \times 10^{-2} \pm 0.11 \times 10^{-2} \\
I_{13} &= -3.79 \times 10^{-3} \pm 0.90 \times 10^{-3} & I_{14} &= 1.64 \times 10^{-3} \pm 0.07 \times 10^{-3} \\
I_{15} &= 1.25 \times 10^{-3} \pm 0.30 \times 10^{-3} & I_{16} &= 1.24 \times 10^{-3} \pm 0.30 \times 10^{-3} \\
I_{17} &= 6.42 \times 10^{-4} \pm 3.00 \times 10^{-4} & I_{18} &= 4.31 \times 10^{-4} \pm 1.30 \times 10^{-4} \\
I_{19} &= 3.00 \times 10^{-4} \pm 1.40 \times 10^{-4} & I_{20} &= -2.02 \times 10^{-4} \pm 8.00 \times 10^{-4} \\
I_{21} &= -1.00 \times 10^{-4} \pm 6.00 \times 10^{-4} & I_{22} &= -5.80 \times 10^{-5} \pm 1.50 \times 10^{-5} \\
I_{23} &= 4.00 \times 10^{-5} \pm 2.00 \times 10^{-5} & & \\
I_{m1} &= 1.14 \pm 0.27 & I_{m2} &= 4.71 \pm 0.54 \\
I_{m3} &= 8.27 \times 10^{-1} \pm 0.93 \times 10^{-1} & I_{m4} &= 2.00 \times 10^{-1} \pm 0.16 \times 10^{-1} \\
I_{m5} &= 1.79 \times 10^{-1} \pm 0.14 \times 10^{-1} & I_{m6} &= 1.93 \times 10^{-1} \pm 0.16 \times 10^{-1} \\
&(Gravitational constants have units of newton meters)
\end{aligned}$$

$$\begin{aligned}
g_1 &= -37.2 \pm 0.05 & g_2 &= -8.44 \pm 0.20 \\
g_3 &= 1.02 \pm 0.50 & g_4 &= 2.49 \times 10^{-1} \pm 0.25 \times 10^{-1} \\
g_5 &= -2.82 \times 10^{-2} \pm 0.56 \times 10^{-2} & &
\end{aligned}$$

**Table A4.** The expressions giving the elements of the kinetic energy matrix.  
(The Abbreviated Expressions have units of kg-m<sup>2</sup>.)

$$\begin{aligned}
a_{11} &= I_{m1} + I_1 + I_3 * CC2 + I_7 * SS23 + I_{10} * SC23 + I_{11} * SC2 \\
&\quad + I_{20} * (SS5 * (SS23 * (1 + CC4) - 1) - 2 * SC23 * C4 * SC5) \\
&\quad + I_{22} * ((1 - 2 * SS23) * C5 - 2 * SC23 * C4 * S5) \\
&\quad + I_{10} * (1 - 2 * SS23) + I_{11} * (1 - 2 * SS2) ; \\
a_{12} &= I_{m1} + I_8 * CC2 + I_9 * C23 + I_{10} * SC23 + I_{11} * SC2 \\
&\quad + I_{20} * (SS5 * (SS23 * (1 + CC4) - 1) - 2 * SC23 * C4 * SC5) \\
&\quad + I_{15} * (SS23 * C5 + SC23 * C4 * S5) \\
&\quad + I_{16} * C2 * (SS23 * C5 + C23 * C4 * S5) \\
&\quad + I_{18} * S4 * S5 + I_{22} * (SC23 * C5 + CC23 * C4 * S5) ; \\
a_{13} &= I_4 * S2 + I_8 * C23 + I_9 * C2 + I_{13} * S23 - I_{15} * C23 * S4 * S5 \\
&\quad + I_{16} * S2 * S4 * S5 + I_{20} * (S23 * C4 * S5 - C23 * C5) \\
&\quad + I_{19} * S23 * SC4 + I_{20} * S4 * (S23 * C4 * CC5 + C23 * SC5) \\
&\quad + I_{22} * S23 * S4 * S5 ; \\
a_{14} &= I_4 * C23 + I_{13} * C23 * S4 * S5 + I_{19} * S23 * SC4 \\
&\quad + I_{18} * C23 * C4 * S5 - C23 * C5 + I_{20} * S23 * S4 * S5 \\
&\quad + I_{20} * S4 * (S23 * C4 * CC5 + C23 * SC5) ; \\
a_{15} &\approx -1.34 \times 10^{-1} * C23 + -3.97 \times 10^{-3} * S23 . \\
a_{16} &= I_{14} * C23 + I_{15} * S23 * C4 * S5 + I_{16} * C2 * C4 * S5 \\
&\quad + I_{18} * C23 * S4 * S5 - C23 * C5 + I_{20} * S23 * S4 * S5 \\
&\quad + I_{20} * S4 * (S23 * C4 * CC5 + C23 * SC5) ; \\
a_{17} &\approx 0 . \\
a_{18} &= I_{15} * S23 * C4 * S5 + I_{16} * C2 * S4 * C5 + I_{17} * S23 * S4 \\
&\quad + I_{18} * (S23 * S5 - C23 * C4 * C5) + I_{20} * C23 * S4 * C5 ; \\
a_{19} &\approx 0 . \\
a_{20} &= I_{18} * (C23 * C5 - S23 * C4 * S5) ; \\
a_{21} &= I_{19} * S23 * S4 * S5 + I_{20} * S23 * C4 * S5 ; \\
a_{22} &= I_{20} * S4 * a_3 * r_{z6} ; \\
a_{23} &= I_{23} * (C23 * C5 - S23 * C4 * S5) ; \quad \approx 0 . \\
a_{24} &= I_{yy5} - I_{zz4} + I_{zz5} - I_{yy5} + m_6 * r_{x2}^2 + I_{zz6} - I_{zz5} ; \\
I_{20} &= I_{yy5} - I_{zz5} - m_6 * r_{z6}^2 + I_{zz6} - I_{zz5} ; \\
I_{21} &= I_{zz4} - I_{yy4} + I_{zz5} - I_{zz5} ; \\
I_{22} &= m_6 * a_3 * r_{z6} ; \\
I_{23} &= I_{zz6} ;
\end{aligned}$$

**Table A5.** The expressions giving the elements of the Coriolis matrix.  
(The Abbreviated Expressions have units of kg-m<sup>2</sup>.)

$$\begin{aligned}
b_{112} &= 2 * \{-I_3 * SC2 + I_5 * C223 + I_7 * SC23 - I_{12} * S223 \\
&\quad + I_{13} * (2 * SC23 * C5 + (1 - 2 * SS23) * C4 * S5)\} ; \\
b_{113} &= 2 * \{I_5 * C23 + I_7 * SC23 - I_{12} * C2 * S23 \\
&\quad + I_{13} * (2 * SC23 * C5 + (1 - 2 * SS23) * C4 * S5) \\
&\quad + I_{16} * C2 * (C23 * C5 - S23 * C4 * S5) + I_{21} * SC23 * CC4 \\
&\quad + I_{20} * ((1 + CC4) * SC23 * SS5 - (1 - 2 * SS23) * C4 * SC5) \\
&\quad + I_{22} * ((1 - 2 * SS23) * C5 - 2 * SC23 * C4 * S5)\} ; \\
b_{114} &= 2 * \{I_5 * C23 + I_7 * SC23 * S4 * S5 - I_{15} * C2 * C23 * S4 * S5 \\
&\quad + I_{18} * C4 * S5 - I_{20} * (S23 * SS5 * SC4 - SC23 * S4 * SC5) \\
&\quad - I_{22} * C23 * S4 * S5 - I_{21} * SS23 * SC4\} ; \\
b_{115} &= 2 * \{I_{20} * (I_5 * C23 * S4 * S5 - I_{15} * C2 * C23 * S4 * S5) \\
&\quad - I_{22} * C23 * S4 * S5 + 8.60 \times 10^{-4} * C4 * S5 \\
&\quad - 2.48 \times 10^{-3} * C2 * C23 * S4 * S5 . \\
b_{116} &= 0 . \\
b_{117} &= 2 * \{-I_5 * S23 + I_{13} * C23 + I_{15} * S23 * S4 * S5 \\
&\quad + I_{18} * (C23 * C4 * S5 + S23 * C5) + I_{20} * S23 * C4 * S5 \\
&\quad + I_{20} * S4 * (S23 * C4 * CC5 + C23 * SC5) ; \\
b_{118} &\approx -2.50 \times 10^{-3} * C23 * S4 * S5 + 1.64 \times 10^{-3} * S23 \\
&\quad + 0.30 \times 10^{-2} * S23 * (1 - 2 * SS4) . \\
b_{119} &= 2 * \{-I_5 * C23 * S4 * C5 + I_{22} * S23 * S4 * C5\} ; \\
b_{120} &= -b_{116} . \quad b_{223} = 0 . \\
b_{121} &= 2 * \{I_{20} * SC4 * SS5 + I_{21} * SC4 * S5 - I_{22} * S4 * S5\} ; \\
b_{122} &\approx 0 . \\
b_{123} &= 2 * \{-I_{15} * S5 + I_{20} * SS4 * SC5 + I_{22} * C4 * C5\} ; \\
b_{124} &\approx -2.50 \times 10^{-3} * S5 . \\
b_{125} &= -b_{116} . \quad b_{323} = 0 . \\
b_{126} &= 2 * \{-I_5 * S23 + I_{13} * C23 + I_{15} * S23 * S4 * S5 \\
&\quad + I_{18} * (C23 * C4 * S5 + S23 * C5) + I_{20} * S23 * C4 * S5 \\
&\quad + I_{20} * S4 * (C23 * C4 * CC5 + C23 * SC5) ; \\
b_{127} &\approx 2.67 \times 10^{-1} * S23 - 7.58 \times 10^{-5} * C23 . \\
b_{128} &= -I_{18} * 2 * S23 * S4 * S5 + I_{19} * S23 * (1 - (2 * SS4)) \\
&\quad + I_{20} * S23 * (1 - 2 * SS4 * CC5) - I_{14} * S23 ; \quad \approx 0 . \\
b_{129} &= I_{17} * C23 * S4 + I_{18} * 2 * (S23 * C4 * C5 + C23 * S5) \\
&\quad + I_{20} * S4 * (C23 * (1 - 2 * SS5) - S23 * C4 * 2 * SC5) ; \\
b_{130} &\approx 0 . \\
b_{131} &= -I_{23} * (S23 * C5 + C23 * C4 * S5) ; \quad \approx 0 . \\
b_{132} &= b_{124} . \quad b_{133} = b_{125} . \quad b_{134} = b_{126} . \\
b_{133} &= I_{15} * S3 + I_6 + I_{12} * C3 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{20} * SS4 * SS5 + I_{21} * SS4 + 2 * (I_{15} * C5 + I_{22} * C4 * S5) ; \\
b_{134} &\approx 0.33 + 3.72 \times 10^{-1} * S3 - 1.10 \times 10^{-2} * C3 . \\
b_{135} &= I_{15} * S3 + I_6 + I_{12} * C3 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{20} * SS4 * SS5 + I_{21} * SS4 + 2 * (I_{15} * C5 + I_{22} * C4 * S5) ; \\
b_{136} &\approx 0 . \\
b_{137} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{138} &\approx 0 . \\
b_{139} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + S3 * C4 * C5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{140} &\approx 0 . \\
b_{141} &= I_{15} * S4 * S5 + I_{16} * C2 * S4 * S5 + I_{17} * S23 * C4 * S5 \\
&\quad + I_{18} * (S23 * C4 * S5 + C23 * C4 * C5) + I_{20} * S23 * C4 * S5 ; \\
b_{142} &\approx 0 . \\
b_{143} &= I_{15} * C4 * S5 + I_{16} * C2 * S4 * S5 + I_{17} * S23 * C4 * S5 \\
&\quad + I_{18} * (S23 * C4 * S5 + C23 * C4 * C5) + I_{20} * S23 * C4 * S5 ; \\
b_{144} &\approx 0 . \\
b_{145} &= I_{15} * C4 * S5 + I_{16} * C2 * S4 * S5 + I_{17} * S23 * C4 * S5 \\
&\quad + I_{18} * (S23 * C4 * S5 + C23 * C4 * C5) + I_{20} * S23 * C4 * S5 ; \\
b_{146} &\approx 0 . \\
b_{147} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) ; \\
b_{148} &\approx 0 . \\
b_{149} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{150} &\approx 0 . \\
b_{151} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{152} &\approx 0 . \\
b_{153} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{154} &\approx 0 . \\
b_{155} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{156} &\approx 0 . \\
b_{157} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{158} &\approx 0 . \\
b_{159} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{160} &\approx 0 . \\
b_{161} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{162} &\approx 0 . \\
b_{163} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{164} &\approx 0 . \\
b_{165} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{166} &\approx 0 . \\
b_{167} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{168} &\approx 0 . \\
b_{169} &= I_{15} * C4 * C5 + I_{16} * (S3 * C5 + C3 * C4 * S5) \\
&\quad + I_{17} * C4 * S5 + I_{20} * S23 * C4 * S5 ; \\
b_{170} &\approx 0 .$$

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Agenda

- Lagrange Method
- Example: Inverted Pendulum
- Example: Cart Pole
- Example: Single Object Dynamics
- Example: Rotor Arms

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Dynamics Example: Grasp

A diagram showing a hand grasping a circular object. Three force vectors are applied to the object: one pointing upwards from the center, one pointing downwards from the center, and one pointing to the right from the center.

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Dynamics Example: Grasp

A diagram showing a hand grasping a circular object with force vectors applied.

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Lagrangian vs. Newton-Euler Methods

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Lagrangian Method

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Generalized Coordinates and Forces

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Generalized Coordinates and Forces

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# **L9: Lagrangian Dynamics**

**Hao Su**

**Spring, 2021**

*The flow and some contents are based on ECE5463 taught at Ohio State University by Prof. Wei Zhang*